

January 3, 1930

A McGraw-Hill Publication

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# AVIATION

*The Oldest American Aeronautical Magazine*

*Inspiring Confidence* IN THE AIR PASSENGER

PRODUCING *Wood Parts* FOR AIRCRAFT

A TECHNICAL DESCRIPTION OF THE *Dornier Do.X*



You  
take  
no  
chance  
with



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# AVIATION

The Oldest American Aeronautical Magazine

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Volume 28, Number 1

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### COMING EVENTS

An interview with Larry Manning, pilot of the Ford plane that landed Earhart this summer, on his observations there. This feature, assisted engineers on the construction of the Stinson plane and lately visited America, who of aeronautical research and airplane design in Russia in an article to be published in an early issue. (Coming next—Continued—Continued—the new Jackson plane.)

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# IN AN AIRPLANE

it's the little  
things that count



The shearing of a bolt . . . the breaking of a pin . . . the failure of a shackle, may result disastrously in the air or on the ground. Those hundreds of obscure detail parts, commonly known as "airplane hardware", are as vital in the question of structural strength as the principal components which lend themselves to ready inspection. Your safeguard is to be sure those "little things" are safe before they are installed in your ship. Aero Supply has acquired that infinite respect for quality gained through years of sympathetic co-ordination with the aircraft manufacturer. All Aero Supply products are built in strict accordance with Army-Navy standards—the recognized criterion of excellence.

**AERO SUPPLY MFG. CO., INC.**  
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SUBSIDIARIES  
NATIONAL STEEL PRODUCTS CO., DAYTON, OHIO  
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IT SEEMS appropriate at this time to define the present position of the Curtiss-Wright Sales Corporation and to outline its plans for the future. The Corporation is now the sole outlet for the entire commercial aircraft production of these outstanding manufacturers in the Curtiss-Wright group. Travel Air Company, makers of one out of every four commercial planes in the United States, builders of 17 types of planes and of the spectacular Mystery ship which outflown every competitive speed plane at the Cleveland Show. Curtiss-Robertson Airplane Manufacturing Company, with its Curtiss-Robin three place cabin monoplane, probably the most widely known moderate priced plane for all-round use. Curtiss Aeroplane and Motor Company, pioneer builders of airplanes and engines, whose Fledgling training plane was chosen in open competition by the Navy and is now widely used by commercial flying schools, whose giant twin-motored Condor is the latest luxury in transcontinental

transport ships. ~ *Moth Aircraft Corporation*, whose DH Moth biplane has a brilliant record of over 10,000,000 miles in the air. Easily housed in a one-car garage, the Moth is destined to be the most popular light plane. ~ *Loening Aeronautical*

*Engineering Corporation*, the world's leading producer of amphibian planes, creators of special types for military, naval and coast guard use, makers of the Air Yacht for transport service and of the Comstar, the luxurious craft now favored by wealthy sportsman pilots. ~ *Keynote Aircraft Corporation*, long noted as producers of large multi-motored aircraft, builders of famous



To build the best airplanes in their class, regardless of cost, will continue to be the policy of the Travel Air Company. This policy, the result of which has been thoroughly proven, has brought the company from a modest beginning in a position of leadership in the aircraft industry to its present position of prominence at Wichita, now ready for production on an increased scale, will continue to serve owners and operators. With a distributor network fortified by able additions and with the country-wide facilities of Curtiss-Wright service stations and dealers now made available to us, Travel Air's position will be further strengthened.

TRAVEL  
AIR  
COMPANY  
■  
■  
■  
VICKERS  
KANSAS

*Malcolm Buch*  
PRESIDENT

AVIATION  
January 4, 1930

CURTISS  
SALES

The Curtiss Robin was widely accepted within a short time because it gives the purchaser more airplane for dollar than any plane in its class. The winning of the world's endurance record and its numerous other accomplishments proved the basic soundness of the type. Now, with the new five up-and-down float that made the largest scale production of the Robin possible, we confidently look for even greater results in 1930. And by having our planes and parts distributed and serviced by the Curtiss-Wright group, every ship will enjoy with it a service guarantee not approached by any competitive line.

CURTIS-  
WRIGHT  
AIRPLANE  
MANUFACTURING  
COMPANY  
■  
ST. LOUIS  
MISSOURI

*Wm. B. Patterson*  
PRESIDENT

AVIATION  
January 4, 1930

WRIGHT  
CORP.

them to the forefront in the industry. Through the unifying control of the Curtiss-Wright Corporation, their position will be strengthened by the coordination of development work, the centralizing of raw material purchases, the pooling of technical information and research data, the elimination of duplication and waste. ~ Backing their program of development is the largest



I am glad that the products of the Curtiss Aeroplane and Motor Company are now to be added into the field of commercial aviation by the Curtiss-Wright Sales Corporation. We hope to serve the public through the new organization as fully as we have served the Army and Navy, where for twenty years Curtiss planes and motors have always done their best.

*Wm. B. Patterson*  
VICE-PRESIDENT

army bombers and of the Patriotic, trimotored 18-passenger transport. ~ These Companies, each pre-eminent in its field, will retain their own identities as manufacturing units, will continue under the same able leadership which has brought

The Moth Aircraft Corporation has become the light plane division of Curtiss-Wright. We will carry on with all distributors and dealers who have handled our products in the past. To these dealers and their customers, will be available over one hundred Curtiss-Wright service stations throughout the country, in addition to existing Moth service in the United States, Canada and foreign countries.

There will be no change in policy, but all contracts for sales will be made through the Curtiss-Wright Sales Corporation, who will represent us throughout the world. I believe that this air-up offers to dealers, as well as Moth owners, a service and an economy which cannot be approached by any other light airplane producer in the United States.

MOH  
AIRCRAFT  
CORPORATION  
■  
■  
■  
LOVELL  
MASS

*Hubert H. H. H.*  
PRESIDENT

aviation enterprise in the world with a capital amounting to one-third the total aviation investment of the country. ☞ Backing their products is a nation-wide service organization including Curtiss-Wright Flying Service, Inc., which operates 40 flying schools capable of training 10,000 pilots a year. ☞ With an army of aircraft covering the entire range of type, style and price, the distributing organization of Curtiss-Wright Sales Corporation faces a world ready and anxious to fly. Aviation activities are now more than doubling every year. Airports, airlines, flying schools, aviation country clubs, are being organized

**CURTISS  
SALES**



*Keynote-Loring Amphibians, originally developed from the military Loring planes, have taken their place as machines of greatest versatility for both private and transport use, as evidenced by their successful acceptance during 1928. To meet the growing demand for these planes, they will hereafter be built at the Broad factory of the Keynote Aircraft Corporation, where augmented facilities will permit large production on an economical scale. The distribution and servicing of Keynote-Loring Amphibians by the extensive Curtiss-Wright organization will insure their practical utility to transport operators and private owners.*

KEYNOTE  
AIRCRAFT  
ENGINEERING  
CORPORATION  
■  
■  
NEW YORK  
CITY

*Alfred H. ...*

VICE  
PRESIDENT

*With an established reputation as leading producers of military aircraft and multi-motored transporters, Keynote's large facilities have now been practically doubled in period the production of commercial planes. With a highly efficient organization, we are now producing various types, including the well known Keynote-Loring Amphibians, searing planes, and a series of Princeton air liners in quantities that will result in manufacturing economies. Distribution will be handled through the nation-wide sales and service organizations of the Curtiss-Wright Corporation, with special advantages to users of our products.*

KEYNOTE  
AIRCRAFT  
CORPORATION  
■  
■  
BOSTON  
MASS.

*Walter H. Beech*  
PRESIDENT

**WRIGHT  
CORP.**

on all sides. Private planes, licensed pilots, air-traveling passengers are rapidly increasing in numbers from month to month. The time for organized sales and service has arrived. Curtiss-Wright Sales Corporation invites all interested in the new era of aviation to meet the executives of the Corporation at the St. Louis Aviation Show, February 15th to 23rd, to inspect its complete line of aircraft exhibited there and to discuss mutually profitable plans for the future. ☞ Complete details regarding the entire line of Curtiss-Wright Airplanes will be gladly supplied from the offices of this Corporation.

WALTER H. BEECH, PRESIDENT



**CURTISS - WRIGHT  
SALES CORPORATION**  
27 WEST 57TH STREET · NEW YORK

# America's Finest Flying Course has been made *even better!*



This Parks Air biplane is completely equipped for night flying with dual instrument, head, landing lights and propeller lights. It is used slightly in the instruction of Transport Pilot students at Parks Air College.



## G IANT tri-motored Ford monoplane, Wasp-powered, added to Parks equipment.

Parks Air College has taken another tremendous step forward.

To its mighty fleet of more than thirty trailing planes America's largest civilian flying school has now added a giant, all-metal Ford monoplane, powered by three 425-horsepower Wasp engines—its exact duplicate of the air liner which is in operation on the largest American passenger lines.

Every Parks transport student will receive thorough instruction in the handling of the new Ford—many hours of its controls, including five hours of absolute mastery of the huge plane.

The addition of this great new ship is one more reason why you should take your training at Parks. No other aviation school has been so progressive in furnishing its students with the very best equipment money can buy.

The Parks training fleet of planes, for both day and night training, includes open cockpit types of several types and carrying power, while planes of latest design, and big transport planes. In addition to the

new Ford monoplane, we have just bought a two-passenger Ryan biplane, powered with a 300 h. p. Wright Whirlwind engine.

Every Parks flying school instructor holds a U. S. Government license, and is a man of high skill and thorough experience. They are all expert in their various fields. Some of them specialize in flying, others are ground school instructors doing nothing but classroom and lecture work, still others are qualified shop men—used as an Mr. Parks personally supervises every activity of the school. He insists upon thoroughness every step of the way. He feels a personal responsibility for the training of every student.

No wonder that Parks Air College was one of the very first schools to be fully approved by the U. S. Dept. of Commerce! Still, wonder that Parks graduates are grabbed up quickly by aviation executives to fill important jobs at big salaries!

You can't afford any training but the best. And you can get the best training only at Parks. Your future is in the balance. Don't gamble with it. There's a big job for you in aviation. Let Parks fit you for it. Take the first step today. Send the coupon for catalog.



Interior view of our new Ford monoplane.



Group of Parks students standing the new tri-motored Ford. Each of these men will be given a thorough course of instruction, including five hours of absolute mastery of this superb plane.



Parks student flying new Ryan biplane—a little after lunch. "Speed of 50 mph"—which has recently been reached by Parks employed the same instructor.

**IMPORTANT ANNOUNCEMENT**—All Parks Air College courses will be advanced in an effective schedule, January 15, 1933. There is still time to enroll at present low prices. Send coupon for more information.

## PARKS AIR COLLEGE

(Division of Detroit Aircraft Corporation)  
151 Missouri Theatre Building

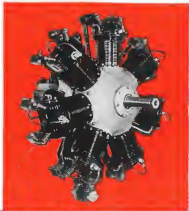
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# Announcing 1930 SERIES WARNER "Scarab" ENGINES



STANDARD ENGINE SPECIFICATIONS COVERED HERE ARE THE  
STANDARD SPECIFICATIONS FOR ALL ENGINES BUILT IN THE  
UNITED STATES IN THE YEAR 1930. THE SPECIFICATIONS  
FOR THE YEAR 1931 WILL BE PUBLISHED IN THE YEAR 1931.  
THESE SPECIFICATIONS ARE THE BASIS FOR THE  
DESIGN OF ALL ENGINES BUILT IN THE UNITED STATES  
IN THE YEAR 1930.

The new 1930 series of 110 H.P. Warner Scarabs, incorporated in which are a veritable wealth of engineering refinements of vital aeroplanist importance, are now ready for your inspection and comparison.

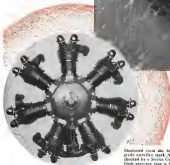
#### Outstanding Features

- Perfection of balance giving smoothest possible operating results.
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- More flying hours with less maintenance.

Warner engines can be secured in the ship of your choice. Complete information on request.

WARNER AIRCRAFT CORPORATION, DETROIT, MICH.

*Finer*  
**GEARS**  
**EXPERTLY**  
**MACHINED**



Engineered from the latest laboratory tested, high-grade materials used, Axelson engine gears are then checked by a Special Constant gear testing machine. Each constant gear is finished in perfect conformity with the Axelson master design. The perfect fit means maximum freedom. Consequently these vital engine parts have an almost indefinitely long life in Axelson Engines 150 h.p., 7-cylinder, Radial Engines—Department of Commerce Approved Certificate No. 14.

The Axelson Company will interest you.

**Axelson Aircraft Engine Co.**  
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Crown Building 21, and 21st Street  
Los Angeles, California  
OF 5-402-3371

**AXELSON AIRPLANE ENGINES**

# MOTH makes Big Flying School Profits



THE GIPSY MOTH'S low first cost; its economy of operation; its folding wings which cut storage cost in half; and its ability to stand up under hard loadings, ensure substantial profits to flying schools. Albany Air Service reports \$1500 net profit from two months' instruction! Let us give you figures showing how MOTH'S will make more money for you.

## Announcement

As Moth Aircraft Corporation is now a unit of the Curtiss-Wright Corporation, Moth airplanes will henceforth be distributed through the Curtiss-Wright Sales Corporation, 27 West 57th Street, New York City.



REGISTERED: THE CURTISS-WRIGHT  
AIRCRAFT CO., INC.

The following schools use Gipsy Moths exclusively for primary instruction:

NEWTON, INC.  
Riverside and West Haven, Conn.  
ALBANY AIR SERVICE  
Albany, N. Y.  
MADISON AIR SERVICE  
Pawcatuck, Conn.  
CUTLER FLYING SERVICE  
Cortland, N. Y.  
AIR SERVICE, INC.  
Albany, N. Y.

**MOTH AIRCRAFT CORPORATION, Lowell, Mass.**  
DIVISION OF CURTISS-WRIGHT CORPORATION  
27 WEST 57TH STREET, NEW YORK CITY

## SKYWAYS INCORPORATED

ESTABLISHED 1911

OPERATING  
FROM SEATTLE, WASH.  
PORTLAND, OREGON  
AND  
WASHOUG, WASH.



THE SEATTLE AIRPORT  
SEATTLE, WASH.

December 26, 1938

Mr. Milton S. Murray, Pres.  
Moth Aircraft Corporation  
Seattle, Wash.

Dear Mr. Murray:

We have just finished our first year of what "Moths" for instruction, and we were very pleasantly surprised to find we had made as large a profit as we had.

We acquired our first "Moth" in April, 1937, and since that time we have been flying four "Moths" daily. We have had 30 passengers since April 1st and have earned \$1,500.00. Students in "Moths" and 3000 flying hours have been logged and 4000 flying hours.

We have at periods of extreme activity repeatedly found these "Moths" were doing better in a "Moth" than we have found in any other airplane we have used instruction.

During this time only one accident has been known and the injury consisted of a small lip, which was in fact a "Moth" accident. The "Moth" which caused it, was a "Moth" of the present time, which was only replaced at about \$100.00. The "Moth" and the accident were.

We feel that these facts should be of interest to those who think that the "Moth" because of its small size, must be too light to stand the strain of student flying.

Respectfully yours,

ALBERT SCHWARTZ

*Albert Schwartz*  
President

cc/

TRANSPORTATION • INSTRUMENTS • EQUIPMENT • AIRLINE PHOTOGRAPHY

## ANNOUNCEMENT

Announcement is made of the consolidation of the Hamilton Aero Manufacturing Company of Milwaukee and the Standard Steel Propeller Corporation of Pittsburgh. The name of the new company, which is a subsidiary of the United Aircraft & Transport Corporation, is the Hamilton Standard Propeller Corporation. Mr. Thomas F. Hamilton is Chairman of the Board of the new company, Mr. Harry A. Kraeling is President and Eugene E. Wilson, formerly Commander U. S. N., Vice President.

★ ★ ★

AN immediate development resulting from this consolidation is the enlargement of the engineering and research departments of the amalgamated companies. In addition to the new plant and other modern facilities at Pittsburgh full use will be made of available speed courses and towing facilities at Milwaukee and Los Angeles to improve the performance of existing designs and develop new types. The Hamilton Standard engineering staff offers complete cooperation with manufacturers in the study and solution of their specific problems.

The consolidated organization also makes possible more complete and rapid service. A modern factory bench and service station has been established at Los Angeles where ample propeller stocks are carried for prompt delivery. Complete heat treating and straightening equipment is being installed to facilitate repair work which will be carried on under expert supervision. Additional branches will be established at strategic points.

For the present, prices will continue in accordance with the price list issued by the Standard Steel Propeller Corporation in January, 1938. Work is now being carried forward on the practical problem of allocating production between various factories. Should manufacturing economies be effected, it will be the policy of this company to reflect such economies by price reduction to the industry.

## HAMILTON STANDARD PROPELLER CORPORATION

Division of the  
UNITED AIRCRAFT AND TRANSPORT CORPORATION  
ATLANTA • MILWAUKEE • LOS ANGELES





# AT EASE IN THE AIR —and on the ground

Theory says that increased simplicity of design is paired by higher efficiency.

Practice says that simplicity of design may give more power than would a slight increase in theoretical efficiency.



OF TWO engines giving equal service in the air, that one is better which can be saved for with least expense and trouble on the ground.

Busy men and women who fly appreciate the Comet aircraft engine because its strength and simplified design give hundreds of flying hours without expert attention.

Any man or woman who can run a grassy gas and wrench can use a Comet for an entire season without fitted help.

Relatively low cylinder pressure and low r. p. m. result in a freedom from strain that increases the life of all wearing parts. Operating without internal strain, Comet gives the train-like regularity of service that is demanded of the family automobile.

**COMET ENGINE CORPORATION**  
MADISON, WISCONSIN, U. S. A.

Comet aircraft engine are manufactured at Madison, Wisconsin, under the direction of the Hixson Machine Company, for 40 years builders of the machine tools in use throughout the world. . . . 7 cylinders, 180 horsepower Hixson's famous roller!



# Leadership

Kelsey-Hayes aircraft landing wheels were the first to be produced in quantities for the industry.

Today the majority of wheels used for aircraft are the Kelsey-Hayes product. In developing new designs and refined construction Kelsey-Hayes Wheels have kept pace with the exacting aeronautical requirements.

Manufacturers have a wide variety of sizes from which to choose, ranging from the 24 x 3 tire wheels to the 36 x 6 axle wheels.

Kelsey-Hayes aircraft engineers will be glad to supply you with data as it applies to designs which you are now developing.

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**KELSEY-HAYES WHEEL CORP.**  
DETROIT, MICH.

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Wing Wheel Corp., of America, Buffalo, N. Y.  
Kelsey Wheel Company, Michigan, Tenn.  
Kelsey Wheel Company, Ltd., Western, Eng.

**KELSEY-HAYES**  
AIRCRAFT LANDING WHEELS

## All Aviation is invited to use the Curtiss-Wright Airports



Berry Airport in Pittsburgh is the gateway between East and West air traffic.

### A nation-wide chain to provide facilities for the entire industry

THE first nation-wide system of major airports in the United States is unified under Curtiss-Wright Airports Corporation ownership and management.

Each port is strategically located to business centers—and on the main highways of both present and future air travel.

Paved areas in for lease today on exceptional terms to transport companies, semi-photographers, railroads and all agencies in air transportation, manufacturers of planes, of engines, supplies, ac-

cessories, fuel companies, and to all commercial operators and private plane owners. Only excellent instruction and best passenger carrying are insured exclusively for the Curtiss-Wright Flying Service.

Among the leading aviation interests who have already leased space on Curtiss-Wright ports are: American Eagle Co., American Merch Distributors, Clifford Dill Jackson, Commercial Air Express, Hollingsworth Aircraft Co., Irving Air Cline Co.,

Inc., Rexair Airplane Motor Co., Madaba Air Lines, Nevada Air Lines, Pennsylvania Air Lines, Polynesian Airways, Inc., Rio Grande Oil Co., Transcontinental Air Transport, Inc., Union Oil Company, Varsity Wing Corporation.

We will be glad to send detailed information about our present airport developments and future plans to executives.

Write to: Department P.T., Curtiss-Wright Airports Corporation, 27 West 27th Street, New York City.



Write to a Curtiss-Wright aviation agent at the Curtiss-Wright Airport.

Let's—these new modern airports offer every facility for plane operation, communication, servicing and repair. A noteworthy feature is the elevated passenger building which aviation contractors a comfortable, sheltered and air conditioned facility.



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in New York Area: Valley Stream, L. I. • Coldwell, N. Y. • North Beach, N. Y. • Hedley Field, N. J. • Los Angeles, Cal.  
in San Francisco Area: Alameda, San Ramon • Redwood, Tenn. • Milwaukee, Wis. • Chicago, Ill. • St. Louis, Mo.  
Cleveland, Ohio • Pittsburgh, Pa. • Baltimore, Md.



The *Monocoupe* and  
**Berryloid**  
AIRCRAFT FINISHES

Mono Aircraft Corporation's first ship was finished 100 per cent with Berryloid as are the planes now coming off the production line. The Red-winged Blackbird inspired the combination on the above Monocoupe—International orange, Boeing yellow and black.

Not  
a complaint  
in  
three years



**MONO AIRCRAFT**  
was founded  
in  
1932  
by  
STINSON  
DINKER  
BYE  
TRAVEL AIR  
EXCHANGE  
DAYTON  
OHIO  
COMMAND AIR  
and  
BIRMINGHAM

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Pittsburgh, Pa.  
Bendure Aeronautical Service, Inc.  
St. Paul, Minn.  
Northrup Airways, Inc.  
St. Paul, Minn.  
Aeronautical Corp. of Canada, Ltd.  
Winnipeg, Canada

## MANUFACTURERS • OF • PROGRESSIVE • AIRCRAFT • FINISHES



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Varnishes Enamels Lacquers  
Beverly Hills, California

# PERFECT SEALING— HIGHER CEILINGS



Valves that stand up longer than the average — without bending, warping or excessive wear — contribute importantly to an engine's reputation for reliability. Perfect valve sealing maintains maximum engine power — affords maximum speed and lift for planes.

Thompson Valves are designed, both metallurgically and structurally, to withstand the extreme stresses of air service. They are used today in 95 per cent of all American-built airplane motors.

THOMPSON PRODUCTS, INCORPORATED  
General Office, Cleveland, Ohio, U. S. A.  
Rochester, Cleveland and Detroit



# Thompson Valves

Original Equipment in 95% of American-Built Aero Motors



## A great vise for the airport

Repairs on motor parts require careful workman ship and proper tools.

With the NuTYP the part is held securely and at the best angle for good work. In the best equipped airport shop in the east, the Empire Airways repair shop, the NuTYP is proving daily its versatility.

The NuTYP vise has been truthfully called the only all around vise in



America, for it gives an unlimited number of holding angles.

The real feature of the NuTYP vise is the two sets of jaws which can be used simultaneously—the mechanism vice with renewable steel jaws and the pipe vise which gives a strong grip without crushing the part held—the cylinder wrench on the side is a patented feature.

The NuTYP vise is made in three sizes, 3 1/2", 4 1/2" and 5 1/2" jaws. All parts are made of first grade steel and carefully inspected.

Your interest in the vise will remove our immediate attention. Write to address below.

The vise shown here stands with maximum capacity of support always at hand.



**NuTYP**  
(new type)  
**WISE**

The Empire Airways shop is one of the best equipped shops in the East.



**OSWEGO TOOL COMPANY**  
OSWEGO, NEW YORK



The "Performing Fifth" . . . John Livingston and his winning WACO. Read for his own story of the Tour and also the WACO "BET" booklet that describes the ship he flew in victory both interesting and useful in your line of change.

**4 1/10 seconds to take  
off . . . 3 1/10 seconds  
to dead stop!**

## The stop-watches click..and WACO sets new records

IT'S the day before the start of The Fifth National Air Tour from Detroit. Ships of all types line the paved runway at Ford Airport, ready for their "unstick" tests . . . or climb aloft, awaiting the signal to test for their "stick" tests.

Down in front of the judges' stand, John Livingston is ready to take off in his WACO. He guns the motor . . . releases the brakes. The wheels move . . . three stop-watches click. And the official timers clock his take-off at the remarkable figure of 4-4/10 seconds . . . from the instant his WACO starts to move until it lifts from the ground!

But his landing time is still more astounding. The same three stop-watches show exactly 3-4/10 seconds . . . from the time the wheels first touch the runway until they come to a dead stop. In actual measured distance, this would be less than 100 feet . . . the size of the standard marking flag that identifies every airport!

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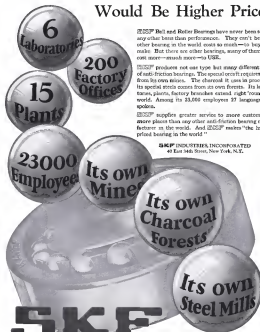
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## AVIATION

THE OLDEST AMERICAN AERONAUTICAL MAGAZINE

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EDWARD P. WARNER, Editor

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### 1930

**T**O PROCLAIM that "the period upon which we are about to enter is the critical one" is so cheap and standardized an editorial trick for attracting attention that we hesitate to employ it. Much honest thought, however, has failed to produce any other phrase in which the significance of the coming year to the airplane industry can be properly defined.

The industry and air transportation have experienced five years of comparatively steady and rapid growth. Within the last three months we have started around a corner, but we have not yet progressed far enough in the morning to make sure where the next section of the road will lead. At no time since the President's Aeronautics Board held its meeting has it been so hard to fathom the immediate future. At no time has it been possible to say with such assurance that the position of the industry three or four years hence will be determined largely by the developments of the next few months.

We shall see, however, that now and the end of the next flying season in the northern states, the public machine to a somewhat extent to bring air transport within the reach of every traveler. We shall see whether or not the business man, not a few hundreds or a few thousands of him but every time that traveler, is prepared to use the airplane under favorable conditions—and if it appears that he is not ready to do so, we ought to have a fairly clear idea of the reason.

Nineteen twenty-nine has been characterized by terrifically intense action. Nineteen thirty will be a year offering more opportunity for maintenance and maintenance. Problems which have so far been met under pressure, upon a basis of expediency and extemporization, will be brought through from their foundations.

The coming year can lay a firm foundation for a sounder growth than any that the industry has yet had, if it is approached in the right spirit. In that spirit there will be no room for strident-like qualities. There must be no evasion of ugly facts. There must be no confusing

of forecasts based upon knowledge with remote dreams based upon hope. The aircraft industry needs no "yes men," but every executive might well include in his own baggage a faculty whose sole duty it would be, whenever he made a statement about the future, to look at him especially and say "Why?" Air transport has a great future, and so has airplane manufacturing, but to get ourselves upon the road to a proper realization we must make 1930, among other things, a year for the suppression of comfortable delusions.

### //

#### Stainless Steels

**T**HAT the great class of ferrous materials, commonly referred to as corrosion-resisting or "stainless steels," have great possibilities in aeronautical applications, has been pointed out and discussed in general terms among the steel and aeronautical trades for quite some time. Despite this general agreement as to potential possibilities, very little of definite engineering study has emerged. While admittedly individual experimentation is proceeding in many widely scattered airplane plants, from the Atlantic to the Pacific, and with the various

**B**EGINNING with this issue, the first page is to be devoted to the contents of the magazine. This change is being made in order that the reader may obtain a more complete record of the editorial action. The title and author's names of the feature articles will be carried, and every editorial department will be indicated. In addition, the contents page will mention feature articles planned for the future.





**T**HE DOX may be considered a monoplane flying boat having superimposed on the main wing an auxiliary surface bearing housings or gondolas for six pairs of engines in tandem. It is also characterized by relatively short stub wings at either side of the hull, this construction being typical of Dornier practice. The engines are Siemens Jupiter nine-cylinder, radial, air-cooled types having 1/2 gear reduction and an aggregate output of 3,300 hp. The Do X has a span of 45 m (147 ft 8 in.) and an overall length of 40.05 m (131 ft 2 in.).

Weights of the craft are shown in Table I which represents a compromise of the actual weights with those estimated by Dr. Dornier at a factory delivered in 1937. A difference in total weight of 2,180 kg. (4,815.8 lb.) or 8.8 per cent is noted on, or the additional weight of the propellers and engines amounting to 702 kg. (1,550.4 lb.) is deducted, a total additional weight of 1,478 kg. (3,265.4 lb.) for the structure of the flying ship, corresponding to 3.6 per cent, is noted for the structure of the craft.

Construction of the Do X took place in the factory of the mode company for Dornier Aircraft, Albstadt. The work was started on December 15, 1937. Launching and first test flights were made on July 12, 1939, providing a total of 370 days for building the craft. A large number of special tools, jigs and equipment were required because of the unusual scale of dimensions. Assembly of the wing, because of the novel construction, was completed in a relatively brief period of time. A great deal of time was spent in the manufacture of templates for construction of the boat panels or frames

The portion lying behind the stern was finished separate from the main hull. The last hull, which forms the backbone of the structure, was placed on an iron beam and then the boat was slatted.

The upper wing was finished complete with engine gondolas and their bearings, as well as the entire set of control rods for the ailerons, and the building in of the tower units. This entire structure was later superimposed on the wing as a whole.

Actual assembly, that is, putting together of wings, boat, steering mechanism and power plant, required 60 days.

During the first flights excessive high oil temperatures were encountered and the temperatures of cylinders of the rear engine were also somewhat high. The first of these difficulties was remedied in a comparatively short time, but the second required considerable maintenance, due mainly to the lack of efficient working temperature governing devices. Among the tests made during the trial flights were investigations of the stresses of the main supporting structural portions of the wing, as well as pressure distribution tests. Observations were made optical of yaw deflections and the stability was investigated with particular care. A series of training experiments also was made.

Flying properties of the Do X are normal and it was found that power-assisted ailerons were unnecessary. The elevator was supercompensated at first. The landing action was smooth normally, but required too great power with a landing gear deflection. This led to increases in compression and diameter of land wheels in the control system.

In Fig. 1 starting periods have been planned against the weight for a large number of trial flights. All of these must take place on Lake Constance, which is more than 400 m. (1,312 ft.) above sea level. Reduced to normal conditions, the curve shows in dashes would result. It must also be considered that the standard Siemens Jupiter engine with compression ratio of 5.3:1 was used. According to specification determined in 1927

## TECHNICAL DETAILS OF THE Dornier X

a safe load of 20 metric tons (44,000 lb.) was guaranteed. Considering present structural weight this corresponds to a take-off weight of 36,000 kg. (79,360 lb.). The graph shows that with a climb on the Lake at Constance the craft required 65 sec. to meet this condition. Reduced to sea level conditions this time would consume 55 sec. The predicted maximum speed was not to be below 200 km. (124 mi.) per hr. with a tolerance of five per cent. Measurements of the performance were taken with water-cooled engines at a height of 420 m. (1,377 ft.) and amounted to 211 km. (130.9 mi.) per hour. Reduced to normal conditions, this amounts to 214 km. (132.8 mi.) per hr., and it is expected that suitable supercharging of engines will result in even greater speed. These tests were made with a set of wooden propellers by throttling the engines to 1,850 r.p.m. the speed at a height of 420 m. (1,377 ft.) was 175 km. (108.4 mi.) per hr.

**T**HE RELATION between range and useful load is shown by taking as a basis an average operating fuel consumption (gondolas and oil) of 270 grams (0.594 lb.) per hp.-hr., assuming a uniform throttling of all engines. The distances in kilometers are plotted as abscissa and the useful loads in kilograms on ordinates. Included lines correspond to take-off weights ranging from 43-52 metric tons (95,380-114,608 lb.). In this graph the additional naval equipment, crew and equipment of passenger cabins was deducted to obtain the net useful load. Naval equipment was assumed constantly to be 360 kg. (793.8 lb.), the weight of the new 1,000 kg. (2,200 lb.) and the weight of equipment indicated by a line

*International interest has been attracted by the Dornier Do X flying boat, largest craft of its type in the world, which was recently completed and test flown in Germany. The present article is a translation and abstract of a paper delivered by Dr. G. Dornier before the Wissenschaftlichen Gesellschaft für Luftfahrt (Society for the Scientific Study of Aircraft). The Do X, as is generally known, represents the most recent stage in the evolution of the Dornier type of design and its development is based on the experience of many years in flying boat design.*

which, at distance zero, corresponds to 3,000 kg. (6,600 lb.) in order to reach the zero point at 4,000 km. (2,485 mi.).

Vertical distances between the equipment lines and the strongly inclined lines corresponding to the different load conditions give the weights available at any time as useful load. It is seen from the diagram that the largest distance that can be traversed in a stop-flight is shown to be about 3,600 km. (2,235 mi.). For pronounced long-distance flight more economical engines would be necessary. Reduction of the present average fuel consumption from 270 to 220 grams per hp.-hr. (0.494 to 0.484 lb.) would entail an increase in the range from 3,600-4,000 km. (2,235-2,485 mi.). A substantial increase in range is expected as soon as it is possible to incorporate a clutch mechanism between propeller and engine, making it possible to stop individual engines singly in emergency flight.

**I**NSURE AS POSSIBLE engineering was eliminated from the beginning of the design work for this great machine restriction was placed on creative possibilities in the design. The self-sufficiency in achieving as closely as possible to conventional standards was given by the fact that, from the moment of the preliminary laboratory



Photograph of the two engine pairs at the port end of the stern shaft. Note the relative thickness of the upper motor and water shaft.



Side view of the Do X showing the relation of wing, hull, and upper motor embracing engine housings.

ness to the final completion, only a few hundred working hours were lost in making changes in the structure.

The preliminary work on the first plan dates back as far as 1924. It was definitely determined in the beginning that a plane of such great load carrying capacity must necessarily be a boat with central hull. The second question of importance was that of mono or multi-plane, and the decision of monoplane was quickly reached. The first plan drawing in dated September 27, 1924, and after certain modifications the final plan was arrived at on June 21st, 1926. Both the boat, 6 m. (19.68 ft.) wide, which was provided for at first, and particularly the hull which was subsequently executed, were carried out experimentally in actual size from wood with complete staining materials.

On December 22, 1926, work was started in the construction department and, with the exception of a change in the shape of the wing struts, only slight changes were made in the original plan.

The third question of importance was that of power distribution. The first design of 1924 provided seven engines totaling 4,320 hp. Twelve engines totaling 3,800 hp. was considered subsequently. Because of the high capacity increase in the capacity of the engines used, the present power plant totals 4,320 hp. The final decision to employ a greater number of engines was arrived at so that engines of lower power and therefore greater reliability could be chosen. Another difficult problem in connection with the power plant was the disposition of engines as well as the decision as to the use of water or air-cooled types. Two years were spent in conducting tests to decide how the engines should be arranged in the hull. The present arrangement was reached as the best compromise under the circumstances. Air-cooled engines were chosen because their use represented a saving in weight of more than 3,000 kg. (6,600 lb.). It was considered that this saving in weight was sufficiently important to offset the relatively smaller fuel consumption of water-cooled engines. The large number of units made tandem installation compulsory.

The reference data for the aerodynamic calculations of the design were obtained by experiments in Göttingen and in the wind tunnel of the Reynolds Aviation Works, and the unusual scale of the models presented definite problems.

As the small wing structure placed above the main

wing has caused considerable discussion, the polar curves are herewith presented. In Fig. 3 the lightly drawn curve of  $\eta$  presents the polar by eliminating the upper wing and the engine gondolas and their supports, thereby corresponding to the ideal condition with power plant completely arranged within the boat or wing and without additional resistance created by the seating of the engines and the support of the propellers. The heavy line  $\eta$  is

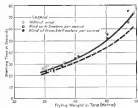


Fig. 3

the polar curve in the present condition and in gliding flight. Dash line C indicates the polar of the present airplane with propellers running as in flight and the dot and dash line D corresponds to the model with gondolas and their bearings, but without the auxiliary surface. The reference area is the surface of the main wing. It is noted, therefore, that, at small angles of attack, the side of the auxiliary surface entails only a small reduction in resistance. To this slight reduction in resistance is added a substantial increase in lift at larger angles of attack. The final curve (heavy  $\eta$ ) is obviously satisfactory in actual practice.

Engines placed in the wing or hull must be cooled and, corresponding to the use of wing radiators, this cooling entails a sacrifice in power and in speed. If propellers are freely arranged, heavy gears are necessary and lead to mechanical uncertainties and additional resistance.

The same applies to driving a propeller by two or more engines. In this case the necessarily large propeller dimensions render it difficult to place the propellers in such a position that they have sufficient protection against splash water. Experience has shown that additional difficulties are encountered in determining the required propeller diameters in a reasonable manner for quantities as high as 4,000 hp. The fact that the arrangement of propellers above the wing greatly enhances the lift is generally known today. A thorough examination

of the possibility of using pusher propellers at the edge of the wing and engines within the wing was made and it was found that the extremely long shafts and difficulties of arranging for their bearings would necessitate additional weight which would offset the aerodynamic gain.

A GREAT DEAL of attention in the investigation period was occupied by questions of structural nature. A full engineer's examination of the wing was refrained from in order to obtain as light a structure as possible. In the choice of the semi-rigid wing construction and in the manufacturing-technical nature were the determining factors.

The wing of the Do X is a three spar truly supported cell. This system provides particular stiffness and resistance to torsion. Furthermore, it affords a guarantee that injury to the wing in case of an emergency seriously the carrying capacity of the entire system.

The wing has a rectangular plan form with slightly rounded tips. The span is 46 m. (157.44 ft.), the depth 9.5 m. (31.16 ft.) and the total projection area, including addition in the upper wing, is 486.2 sq. m. (5231.41 sq. ft.).

The weight of the wing complete with stiffeners, upper wing structure and all bracing is 7,280 kg. (16,029.8 lb.), the unit weight of the wing being 15.5 kg. per sq. meter (3.16 lb. per sq. ft.). The wing with a total weight of 60 metric tons (132,400 lb.) corresponds to the DVL regulations covering passenger aeroplanes in which an  $\eta$  factor is in force at the present time.

The central spar lies approximately in the position of maximum profile height. Front and rear spars are moved far apart, their distance from the central spar being 2.8 m. (9.18 ft.). The distance between the two outer cross members is as high as 3.6 m. (11.80 ft.).

The upper wing is not used for relieving the main supporting surface and, strictly considered, it merely serves for stiffening the engine bearers. With the exception of some small fittings the entire wing frame work is made of aluminum alloy. This material is chosen because of the difficulty in securing appropriate sections in the required dimensions.

A cross-section and a view of a part of the central spar at the connecting point of the strut are shown among the accompanying illustrations. Flanges of the spars are made of pressed aluminum alloy angles and bearless simple as those used in bridge building and additional benefits or segments are provided as required.

At points of lower force loading the vertical webs of the angles are added out for purposes of decreasing the weight. In construction the front and rear spars are similar to the central spars. At this point it is of interest to note an experience that occurs in increasing dimensions not only of spars but of other structural members. This is the decrease of the amount of work expended per unit of finished construction weight with the increase of dimensions of structural members. The determining factors in expenditure of work are chiefly the number of joint points and the number of rivets. In the spars of the Do X there are an average of 2.5 joint points per running meter, the corresponding numbers in the Dornier Doppeldecker being 3.3 and in the He 59 3.5. In the Do X there are 9.8 rivets per kilogram of finished spar weight, while in the Saperow they are 3.3 and in the Wal 44.

Many months were expended in determining the ultimately adapted spar construction. Special experimental apparatus was designed particularly for this purpose. On account of the unusually great test load to which the

spars were subjected rigorous foundation work was necessary to render the experimental apparatus undisturbable. The customary method of loading the spar with sand bags or iron ballast was discarded and it was decided to arrange receptacles filled with water at points of unaccounted loading. This method gave good results. Particular attention was paid to the conduct of the spar when the elastic rings and transmitters were employed to ascertain the tension in various structural members.

Control measurements of the tension in structural members of the central spar were conducted with a Stach instrument placed at the disposal of the Dornier Company by the German Experimental Institute for Aviation, which organization also co-operated in the use of this instrument. Fig. 4 shows the deflection measured and calculated for the central spar at different load stages.

The cross members are also made of pressed profiles in the main. At their point of connection with the first spar it was partly necessary to build in frames in order to facilitate passage through the wings. The deflection caused by cutting down the cross members averaged at distances of from 2.8 to 3.6 m. (9.18 to 11.80 ft.) with

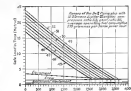


Fig. 4

the three spars are covered by plates resistant to bending and covered with sheet metal. Flanges of the spars have been called "wing skin fields". The portion of the wing lying behind the rear spar is constructed as an independent deck. The nose is made entirely of metal and equipped for stiffening or bracing the front spar against bending. The wing skin fields are simply and easily made and easily assembled. Their attachment to the main chord is assisted by crossmembers with bolts lying entirely within the wing. Although the wing has not as exceptionally heavy profile, the height of the spars, due to the great dimensions of the oval, is so substantial that even in flight it is possible to walk on nearly all parts of the wing.

As previously mentioned, the upper wing serves for a structural stiffening or bracing of the engine bearers and is made entirely of metal. It turned out to be relatively heavy, since its own weight amounts to 18 kg. per sq. m. (3.98 lb. per sq. ft.) in reducing deformation and possible vibration to a minimum, too much apparently has been done in this case. The unit weight of the total wing previously mentioned as 15.5 kg. per sq. m. (3.17 lb. per sq. ft.) includes the weight of the upper wing. This measure weighs, therefore, affects the total unit weight unfavorably. It is desired to eliminate loadings

Table 1. A Comparison of the Actual and Estimated Weights of the Do X

Construction Group	Actual Weight (in kg.)		Estimated Weight (in kg.)		Difference between Actual and Estimated Weights in Columns 3 and 4	
	kg.	lb.	kg.	lb.	kg.	lb.
1. Wing with support	10,000	22,046	10,000	22,046	0	0
2. Landing gear	1,000	2,204	1,000	2,204	0	0
3. Engines	4,320	9,525	4,320	9,525	0	0
4. Fuel	2,000	4,409	2,000	4,409	0	0
5. Oil	1,000	2,204	1,000	2,204	0	0
6. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
7. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
8. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
9. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
10. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
11. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
12. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
13. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
14. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
15. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
16. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
17. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
18. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
19. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
20. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
21. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
22. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
23. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
24. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
25. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
26. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
27. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
28. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
29. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
30. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
31. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
32. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
33. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
34. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
35. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
36. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
37. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
38. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
39. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
40. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
41. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
42. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
43. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
44. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
45. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
46. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
47. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
48. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
49. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
50. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
51. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
52. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
53. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
54. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
55. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
56. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
57. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
58. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
59. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
60. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
61. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
62. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
63. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
64. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
65. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
66. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
67. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
68. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
69. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
70. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
71. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
72. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
73. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
74. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
75. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
76. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
77. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
78. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
79. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
80. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
81. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
82. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
83. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
84. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
85. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
86. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
87. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
88. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
89. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
90. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
91. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
92. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
93. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
94. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
95. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
96. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
97. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
98. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
99. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
100. Landing gear with axle	1,000	2,204	1,000	2,204	0	0
Total to be	34,940	77,018	34,940	77,018	0	0

able additional strain on the main system the ends of the upper wing are displacably joined to the central portion between the two outer gondolas. The engine bearers are a frame work construction and sketched in schematic fashion. For purposes of aerodynamic forms are provided at both ends.

AS PREVIOUSLY MENTIONED, the decision that the craft must be a boat with central hull was made only in the design work. In the first place, it was considered desirable to achieve the necessary stability without resort-

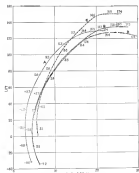


Fig. 1

ing to supplementary displacement bodies. Experiments indicated, however, that this was not possible on the scale under consideration without involving important disadvantages. The designers, therefore, resorted to the idea of wing stubs which had proved so successful during a long period of experience in Dornier flying boat designs.

The hull area below the water, in as far as they affect the steering process, were but slight modifications of these used in former Dornier craft. The central longitudinal step is retained and it horizontal movement in the direction of flight at its rear end, pointing into a slight "V" shape toward the front. The purpose of the hull below lying at the sides of the longitudinal step are given a slightly concave shape. The fore-body is broken parabolically in portions which at the start lie in quiet water. At the places where the struts penetrate the hull they are rounded off strongly cross-wise to the direction of flight. The height of the struts at the root, therefore, became substantially greater, making possible a central cross more rapid construction of supporting members. At the start the water is accelerated outward, far to the arch as shape while in the former design it was

compressed in the corner between the under side of the web and the side wall of the boat.

In Fig. 5 at the right-hand side a diagrammatic cross-section of the main step of the Do X is shown at the left for purposes of comparison with that of a corresponding section through the Dornier Wal at the right. It is noteworthy that the ratio of the width  $B$  of the boat measured over the struts decreases to the width  $b$  of the hull proper as the area increases. While this ratio is about 2.80 in the smallest Dornier boat (the "Lalotte" which has a take-off weight of about 570 kg.) (1,274 lb.), it drops to 2.40 in the Wal type and 2.15 in the Do X flying boat. In the future boats of about 100 metric tons (220,000 lb.) it will be possible to employ struts of a rudimentary character and in boats of still larger dimensions, the struts can be eliminated altogether.

An essential characteristic of the hull design of the Do X is the division into three independent decks which has been carried out for the first time in aircraft in this design. The uppermost deck is reserved for the commander and contains accommodations for him and for the pilot as well as a control room and rooms for radio and auxiliary apparatus. The next deck is reserved exclusively for passengers and is 23.5 m. (77.08 ft.) long, 2 m. (6.56 ft.) high and 3.5 m. (11.48 ft.) wide at the widest place. The lowest deck contains operating mechanism and space for provisions, supplies, freight and baggage.

The overall length of the boat is 40.05 m. (131.36 ft.) and the width measured over the struts is 10 m. (32.80 ft.). The width exclusive of the struts is 3.5 m. (11.48 ft.). The maximum height is 6.4 m. (21 ft.), while the draft empty is 0.8 m. (2.62 ft.) and with 30 metric tons (110,000 lb.) is 1.05 meter (3.44 ft.). With 30 metric tons (110,000 lb.) the waterline height is 4.88 meters (15.99 ft.). Including the struts the boat has a total volume of 400 cu m. (14,125 cu ft.). This is four times as great as the volume of the Sepered.

The main spar element cross-section without struts has a content of 12.2 sq m. (188.8 sq ft.). The panel distance is 0.7 m. (2.29 ft.). A total of 58 cross panels is used. One of the innovations in the structure is a lead bar 23.5 m. (76.5 ft.) in length running from the bow to the end of the stern step and having a maximum height of 2.12 meters (6.95 ft.) which produces a substantial stiffness of the boat. Keelsons are arranged at the right and left of the lead bar at distances of 0.9

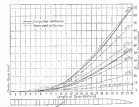


Fig. 4

and 1.58 m. (2.95 and 5.19 ft.). This in conjunction with the cross panels produces a high efficiency compound effect. Throughout the range of the boat bottom which is directly exposed to the stress produced by waves the exceptionally strong bracing struts are divided into fields of about 0.63 m. (2.06 ft.) by crossing transverse and longitudinal panels.

In the construction of the cross panels an attempt was made wherever possible, considering the weight available, to make use of pressed profiles. It is interesting to note at this point that the weight per cubic meter of boat volume in the "Lalotte" is 299 kg. per cu m. (1.86 lb. per cu ft.), in the Wal is 26.2 kg. per cu m. (1.63 lb. per cu ft.) and in the Do X is 21 kg. per cu m. (1.3 lb. per cu ft.). In spite of this fact the amount in the material, assuming equal loads, are substantially less in the Do X than they are in the smaller boats. The larger craft, in so far as local stresses are concerned, is not in a nearly more fortunate position than the smaller ones in which the small wall thicknesses of the sheet metal and profiles used are more susceptible to the danger of local stress and spalls. Both hull and struts are subdivided into a large number of bulkheads, the hull having nine water-tight compartments and each strut having four. The reserve displacement is exceptionally large. Displacement of the struts alone amounts to 43.5 cu m. (1,534 cu ft.).

AREA of the horizontal tail surfaces amounts to 53.4 sq m. (579.5 sq ft.), while that of vertical surfaces is 19.0 sq m. (204.4 sq ft.). All rudders are balanced by auxiliary surfaces. Elevator clearance above the water line is 6 m. (19.68 ft.) and clearance of the fin-plates to the rear protects the rudder against the jet of the water. Rudders are actuated by traction rods suspended from pushdown levers. The rods are secured entirely in ball bearings.

Trimming balances are built into the surfaces, the trimming being accomplished by setting the angle of attack of the corresponding compensating surfaces. This insures the economy of employing power of any kind, and the control can be operated effectively from the pilot's seat. A portion of the commander's deck intended for the pilot is shown in one of the accompanying photographs. The two small laterally disposed hand wheels are used for trimming and the large hand wheel serves to actuate the outer rudder. Manual wheel steering is employed.

THE TWELVE ENGINEERS are rated at 825 each. They are twelve, *air-minded*, types having a gear ratio of 1:2. Special attention is given to the design of the engine bearers, the structural members entering into their construction being especially large dimensions. The preliminary investigation of these craft included assessment of the stresses in an engine mounting subject to combined thrust and turning moment as well as the weight of the engine.

One of the important features in the Do X is the fact that, probably for the first time in aircraft history, the pilot has been relieved of the task of supervising the power plant. Of course the pilot has all of the engines under an immediate control if necessary. This is accomplished by means of two levers lying directly beside each other and which can be operated jointly by a single grip. Operation of one of the two levers provides the possibility of thrusting the starboard or port group of engines, thus being particularly useful for maneuvering

on the water. All engine cylinders can be cut out simultaneously from the pilot's seat.

Two collective revolution counters and an electric accelerometer apparatus are included among the other engine instruments. The accelerometer device consists of twelve small lamps, each arranged to an engine and so

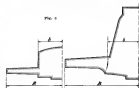


Fig. 3

connected that they light when the engine is stopped. Complete separation of engine supervision from the pilot is produced in all other matters.

Starting, long distance supervision and control stopping of the engines are handled from the control or switch room. Connection between the control room and the engine gondolas is made through a tunnel leading through the wing. Engines are started by compressed air which is switched by a small controller operated by a separate gasoline engine which also serves to produce the entire electric current for the aircraft.

THE fuel system in general is similar to that used in the former Dornier planes. In the operating load deck actually four cylindrical receptacles of 3,000 l. (792 gal.) capacity each, as well as four additional receptacles of 1,000 l. (264 gal.) capacity each, are provided, permitting a total gasoline capacity of 16,000 l. (4,224 gal.) space. These tanks are placed directly on the bottom of the boat and communicate with a so-called relieving pot lower, which has a tank by means of which it plugs in two receptacles, each having a capacity of 300 l. (79.2 gal.) space in the wing space. In order to provide a high degree of safety in the fuel system, gasoline may be pumped in three different ways, i.e., by a pump with windmill drive, an electrically operated pump or a hand pump. On the nose tank the gasoline is fed to the carburetor, also by means of pumps, and excess fuel flows back into the collecting pot. Sight feed gauges on the commander's deck permit control of the fuel supply. Typing is easily accessible and can be reached at all times, even when the plane is in flight. Oil tanks in the wing have an aggregate capacity of 1,600 l. (4,224 gal.).

The specifications of the Do X are as follows:

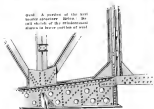
Wing Span	48 m. (157.8 ft.)
Overall Length	40.05 m. (131.25 ft.)
Weight Empty	29,914 kg. (66,404 lb.)
Wing Load	20,000 kg. (44,000 lb.)
Maximum Speed (Sea Level)	214 km. (132.8 m.p.h.)
Cruising Speed	1,850 m.p.h. (1,377 ft.)
Power Plants	12 Siemens Jupiter, 525 hp. each.
Total Power Plants	6,300 hp.



Note: The body adopted plan is based on the construction of the Do. X.

## CONSTRUCTION DETAILS OF THE *Do. X* FLYING BOAT

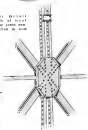
Below: Photograph of the hull structure during the early stages of assembly at the factory. Right: Detail showing all the girders which in the lower hull form corner at each corner.



Good: A portion of the hull structure during the early stages of assembly at the factory. Right: Detail showing all the girders which in the lower hull form corner at each corner.



Water: Detail showing all the girders which in the lower hull form corner at each corner.

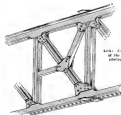


Below: Assembly of the power plant.



Below: Detail showing all the girders which in the lower hull form corner at each corner.

Water: Detail showing all the girders which in the lower hull form corner at each corner.



Water: Detail showing all the girders which in the lower hull form corner at each corner.

Water: Detail showing all the girders which in the lower hull form corner at each corner.



Water: Detail showing all the girders which in the lower hull form corner at each corner.





# DEVELOPMENT OF THE "M" Wing Sections

By MAX M. MUNK

P.O. Dr. Eng.  
Alexander Lehmann, Chiefly Strong

## A Discussion of the Results of a Systematic Series of Model Tests of Mathematically Derived Wing Sections

THE "M" SECTIONS constitute a family of wing sections the aerodynamic characteristics and shape of which were published at first in 1906 [Max M. Munk: Model tests with a systematic series of 27 wing sections at full Reynolds number, *NACA, Tech. Rept. No. 221*]. Their shapes were varied systematically in order to find good ones and to learn about the influence of the variables. It was the author's first attempt at proceeding in such a way, for the Goettingen sections follow no strict system. Several airplane designers have adopted the outstanding "M" wing sections since then, with good results. This seems to indicate that it is still possible to improve the existing wing sections, for if we had already exhausted the possibilities, the first attempt of a systematic exploration would not have yielded sections that were immediately adopted by the industry. It seems further to indicate that we were on the right track in the layout of the first systematic series of wing sections designed for experimental exploration, and that we should continue along the same line. Further attempts to improve wing sections by systematic tests did not find support when this first series was formed. It is the purpose of the following article to create interest in such further exploration and to explain the principles adopted for the layout of the first series of the "M" sections to serve as a guide for further research along that line.

The "M" wing sections are so-called mathematical wing sections. There are two classes of such curves. The first and older class, the elliptic sections, all of which are the Joukowski sections, are curves not merely determined by mathematical rules, but furthermore determined in such a special way as to make it possible (though not exactly easy) to draw the theoretical streamlines created by such sections and to compute the theoretical air forces. They are examples of mathematically good ones or better than the purely empirical curves, drawn by the designer from experience and artistic feeling. The "M" sections subordinate themselves to this class of mathematical sections. They are themselves simple mathematical curves, but they are as the constants for simple mathematical equations. Aerodynamic viewpoints contributed to their shape in quite another way. M-1 to M-27 were intended to be the first series of a more ambitious program. For the purpose of systematic exploration all wing sections were classified according to their thickness, their effective camber and

of theoretical flow was always strictly aerodynamic, and was further developed in 1923 by the author's discovery of a simple mathematical method for computing the theoretical aerodynamic characteristics of all wing



Fig. 1—Standard camber line

sections, mathematical or not and with or without mathematical streamlines. We shall come back to this method, because it has also been used in the layout of the "M" sections.

The other class of mathematical sections, those without the background of special methods for computing the theoretical flow, are drawn by the guidance of simple mathematical rules or formulas, for practical reasons only. This method of layout assures the greatest possible continuity of the contour, and thereby improves the aerodynamic properties. Furthermore, it makes it easy to obtain families of sections which vary systematically, and for which the results of the tests can easily be classified, and intermediate sections drawn by interpolation. The fact that for every empirical section a curve determined by mathematical equations can be found so as to agree practically with it, shows sufficiently that the mathematical sections are not in such measurably good ones or better than the purely empirical curves, drawn by the designer from experience and artistic feeling. The "M" sections subordinate themselves to this class of mathematical sections. They are themselves simple mathematical curves, but they are as the constants for simple mathematical equations. Aerodynamic viewpoints contributed to their shape in quite another way. M-1 to M-27 were intended to be the first series of a more ambitious program. For the purpose of systematic exploration all wing sections were classified according to their thickness, their effective camber and

their effective S-shape. Subclassifications were left to the future. The thickness is required for structural reasons, and affects the aerodynamic properties of the section considerably. The effective camber and S-shape are not limited by structural considerations, but are fully at the disposal of the designer for influencing the aerodynamic properties to the best advantage. A large thickness increases the minimum profile drag, but may raise the maximum lift and speed out the range of a fairly good performance. The camber and S-shape have to be considered together. The ratio of their magnitudes determines the travel of the center of pressure, also called the stability of the wing section. The camber by itself increases the maximum lift and the lift for maximum profile drag, but also increases the travel of the center of pressure. The S-shape diminishes this travel, but also the maximum lift and increases the minimum profile drag. The problem is to find the best combinations of these three characteristics, giving the best compromise. Wind tunnel tests at full scale Reynolds number were accordingly started with sections embracing three very simple mathematical curves chosen to represent each of the three mentioned general geometrical characteristics of the sections. The curves were selected primarily from the experience of former wind tunnel research in connection with a simple mathematical formula. Their shape was influenced, however, by the nature of certain integrations, which have to be per-

formed on the "M" sections, it is necessary to review in a few words the principles of the general wing section theory. A wing or wing section, now considering plane flow only, is a deformed airfoil. Since air deflection is the essence of a wing, an airfoil deflecting air toward its down chord is now creating a lifting force, thereby performing the duty of a wing. This deflection takes place along the whole length of the wing section along its whole chord, but not with equal intensity at all



Fig. 2—Standard thickness curve reduced to zero-thickness

points. The intensity of the deflecting process depends on the direction in which the wing section extends at each point, in connection with its position relative to the leading and trailing edges, the wing section elements in the rear being better able to impress their direction on the airflow than those near the leading edge.

The condition at the leading edge and the trailing edge are of particular interest. Observation has shown that, unless "berthing" has occurred, the air is always actually deflected to flow parallel to the rear edge on both its sides, upper and lower. Not so at the front edge. The air arriving there is just beginning to be subjected to the deflecting action of the wing, and the result is still imperfect. At all but one angle of attack, where the conditions happen to be favorable, the air fails to strike the section at the leading edge and to flow parallel to the forward portion of the section on the upper and lower side of its nose. It rather strikes sideways and flows around the nose, either from bottom to top, at large angles of attack, or from top to bottom, at small ones. A smooth leading edge facilitates this flow around the nose, and thus spreads out the range of a fairly small profile drag. It increases, however,



Fig. 3—Standard S-shape

formed with the airfoils of the sections in order to compare their theoretical aerodynamic properties. The shape of the curves makes these integrations relatively simple. This was not, however, considered a point of prime importance, but it happened that such shapes also looked promising with respect to their aerodynamics.

Before proceeding to discuss the three curves which

the minimum profile drag, when the air strikes the leading edge directly in front and hence no flow around takes place. At the one angle of attack, the air flows parallel to both leading and trailing edges, and hence parallel to the trailing edge only as it does at all other angles. This is therefore the angle of attack of the most perfect condition, the air flowing parallel to the section along its entire length. The section would be designed, in that case, not to be used at any angle of attack only as in the case of certain thin blades, but fundamentally this is not true of airplane wings.

The thickness of the wing section at its different stations along the chord has a decided influence on the choice of the roundness of the leading edge, and it further influences the profile drag and the moments felt, that is the lift as well as the heeling sets on. The thickness is of secondary importance to the deflecting action.



the distance of the station from the center. It is drawn in Fig. 2 for an inclination of 45 deg. at the center, the tangent of this angle being 1. The tangent of the inclinations at the ends is twice that large. Thus a symmetrical S-curve rather than a curve of constant S effect. It has also actual effective camber. The angle of effective camber of the parabola is equal to the angle between the chord and the line of connection between the trailing edge and the maximum camber at 50 per cent chord. Its S-effect is zero. The S curve in Fig. 1, with 45 deg. inclination at the center, has half the effective angle of camber than the parabola with 50 per cent camber in the same figure. Assuming the right side to be the trailing edge, the camber of the S-curve is negative. Its angle of effective S-effect is half of its angle of camber, but positive.

By a suitable combination of these two curves a curve of fixed center of pressure can be obtained. One unit parabola (50 per cent camber) combined with two unit S curves (45 deg. inclination) gives a pure S curve, having S effect but no camber effect. These unit parabolas and four unit S-curves combine to produce a curve of equal camber and S effect, that is a curve of constant center of pressure. Otherwise stated, the ordinates of the S curve in Fig. 1 have to be multiplied by a factor 4/3 times as large than the factor for the parabola in order to obtain a curve of a theoretically fixed center of pressure. For the design of the "M" sections, several combinations were used, the ratio of the factors varying around 1.33, between 1.00 and 1.50.

The thickest curve at last was obtained from the ordinates of a parabola through the leading edge with the chord as horizontal axis. These ordinates were multiplied by the distance from the trailing chord. The unit thickness curve, then ordinates reduced to one half, the maximum thickness being then 54-632 per cent of the chord, is shown in Fig. 3. The maximum thickness of the unit thickness curve is 108.667 per cent and is located at 33-33 per cent of the chord. The three basic curves are thus seen to be chosen in very simple algebraic ratios. We did not, in spite of certain advantages, take nine curves for the camber and S curves, developing the chord in 90 deg. and 360 deg. and plotting the rise. The integrations for the effective cambers are then made difficult and, moreover, it is not possible to obtain a simple thickness curve that was.

M-1 to M-5 are obtained directly from the thickness curve, by multiplying it by suitable factors in order to reduce the thickness to different maximum values. It was, however, necessary to add a third thickness near the trailing edge, in order to avoid too sharp and vulnerable an edge. These three thickness curves were first combined with two camber curves, obtained in turn by combining one standard S curve with either two-thirds of the ordinates of the standard parabola or with one times its ordinates. One tenth of these ordinates were taken as fundamental camber lines, designated by "a" and "b". These were then again combined by doubling them, adding each curve to double the other one and by doubling both and adding them to each other. In this way eight camber curves are obtained, the combinations having an even greater theoretical travel of the center of pressure than the original camber curves, "a" and "b". The tests best out that actually gave sections of small or zero travel were discarded.

The first 27 M-sections constitute therefore an exploration in the region of small center of pressure travel.

A small travel is desired by most designers, although the advantage is not decisive, and there is in particular no advantage in having the travel entirely eliminated. It is, however, a fact, that sections with straight leading edges (zero sections) show a remarkably small maximum profile drag, but are deficient in maximum lift. It appeared, therefore, promising to try to increase this maximum lift without impairing the maximum drag maintaining the absence of a large travel of the center of pressure, by adding some S and camber neutralizing each other almost or fully. This possibility suggested itself, furthermore, because this range of combinations was little explored at that time. The results have justified the expectations. It appeared that a small camber and S gave good results, and that small variations of the ratio of the camber to the S did not affect the result much. The best sections were the thickest ones combined with the smallest cambers, M5 and M12. Their plans are reproduced in Figs. 4 and 5 from the original paper.

The author is now about to start an investigation of further M-sections, based on the results obtained with the first series, and directing the further development in more intimate touch with practical needs than he had while associated with a pure research organization. The Alexander Aeronauts Company have made appreciations for a wind tunnel to assist in the improvement of their products, and it has been decided to study all questions, whenever it seems promising, in systematic flight work. The author hopes to be able in a near future to lay before the readers results with more, better and up to date M sections, adapted to the last needs of the designer, and obtained in wind tunnel and in actual flight work.

#### Appendix

For the convenience of the reader the relations discussed in the preceding articles are repeated in explicit form.

Let the chord extend from  $x = -1$  to  $x = 1$ , this latter point being the trailing edge, and let  $\delta$  denote the ordinates at the section curve.

Parabola Ordinate	Equation for $\delta$ at $x$	Angle of Camber at $x$	Angle of Chord at $x$	Max- imum Order of Inclination at $x$
Unit parabola	$\delta = 1 - x^2$	0	0	300 or 1000 at $x = -1$

Thickness I =  $0.85 T$ , Thickness II =  $0.673 T$ , Thickness III =  $0.11 T$ ,  $a = 0.1 (S + 1/10)$ ,  $b = 0.1 (S + 2)$  Camber integral

$$\text{Camber integral} = \pi = \frac{1}{2} \int_{-1}^1 \frac{\delta dx}{(1+x)\sqrt{1-x^2}}$$

$$\text{Moment integral} = m = -\frac{1}{2} \int_{-1}^1 \frac{\delta x dx}{\sqrt{1-x^2}}$$

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## THE IMPORTANT MATTER OF Confidence AND THE AIR PASSENGER

*Some Straight-From-the-Shoulder Criticism of the Highly  
Unprofitable Methods of Handling Passengers  
Now Being Followed by Some  
Transport Companies*

By LIEUT. COMDR. FRANK W. WEAD, U.S.N. (Rtd.)

THE FUTURE of air transportation in the United States depends entirely upon the amount of air passenger traffic that can be developed. Freight, mail, express, sport flying, the various government branches of aviation, and private air traffic will always help to maintain an aeronautical industry. Such activities have, however, done practically nothing toward developing the network of excellent airways that now covers Europe, and without doubt we will ultimately find that the same fact is true in our own country. The air passenger, then, should be entitled to every consideration that can possibly be given to him. The mere fact that airplane travel has so long been inefficient. The intelligent air passenger not only demands that he travel fast, but in addition, that he travel with convenience, with comfort and with safety.

After some years of active flying the writer of necessity drifted into other channels. Many of my associates became average men and women to whom flying was only a dangerous, thrilling pastime indulged in by individuals that were slightly unbalanced. Lindbergh flew to Paris; air transport came along with a bang, after splendid flights made first overhauled, made failures; the public became interested. People were told on television. They wanted to learn about it. They wanted to fly and they did fly. I saw the start of the air transport game, not as a participant, but from

a predestined seat, and, more recently, as a customer. My friends constantly asked, and still do, my advice on aviation matters, questions on requirements, types of planes, schools, transport lines, regions, and every other thing under the sun connected with flying. Some twenty-odd of them took an instructor and all have a private license or better at this time. Others wanted to see the airplane for travel, for business, for pleasure. Some of these latter have been well pleased with their experience and travel by air at every opportunity. Others had not been so fortunate.

GIVEN NUMEROUS ONE of these air passenger friends of mine took a very short trip over a scheduled airway that handles considerable traffic. They got into their seats easily and without delay, took off and climbed to 1500 feet. Their consciousness died and they began to enjoy the trip. While still some miles from their destination the pilot without warning cut his plane, showed the nose down, and lost a thousand feet of altitude in a steep dive. They were spaced his throttles and continued on his way. His dive was unexpected. The passengers had to brace themselves to avoid being thrown out of their seats and they thought that something was wrong with the plane or with the pilot. They were frightened. They were intelligent people of large earning capacity. They were keen nothing about aviation, and they were

on their first flight. Perhaps the pilot only wanted to give them a thrill. Perhaps he did not think of his passengers or was careless. Such people don't want airplanes to thrill them any more. This has just passed. They want airplanes to take them some place. The curiosity of these particular passengers about aviation suddenly died. They took up their return tickets and came back home in other means of transportation. There was a little sadness of the fact that they did not like to fly, but they weren't having any more of it. Properly handled, they might have developed into enthusiastic lovers for air transport. A lot of our friends were curious, non-flying planned trips by air. But most of them had been talked out of it—one unnecessary dive did the work.

GROUP two were off for a well-earned jaunt both ways by air. They sent a messenger down to the office of the air line to make reservations. He paid the charges and returned with reservation slips. The following day at the airport, one of the party approached the man in charge and handed him the slips. Instead, the slips had to be taken to the ticket window and exchanged for tickets. Each member of the party had to personally confer with the ticket agent and give him full information as to age, race, address of all relatives, bank balance and previous condition of service. Is return he or she received a ticket several yards long, for a long ride, and a considerable amount of baggage. The party then placed these while they one by one accompanied that vital piece of baggage. The plane was 15 min late leaving off. On the return trip at the other terminal of the line they waited 35 min. for an overdue plane to arrive. No information was available as to when it might get in. There was no one to ask about it, but they decided to sit. There were more but they didn't like to sit on a freshly filled flying field. From a standpoint of flying equipment and competent pilots, this particular airline, to my mind, leaves little to be desired. They are, however, neglecting the thing that is most vital to them. They are neglecting the people who fly. They have made a mistake. Some information regarding passengers should perhaps be recorded. It should be kept at a minimum and handled expeditiously. When people travel to save time they should not be required to waste it being tickled or waiting for airplanes.

GROUP two were off for a well-earned jaunt both ways by air and were assigned seats in plane number six, 17 for a half-hour exhibition flight over the water. They were seated there. They were to delay at the airport. Some of them had important engagements—others were on their lunch hour. At the airport they saw plane number 17 make four fully loaded trips before they finally got on board after a three hour wait (if that isn't down they lost their place in line) and finally several plane loads of people who arrived at the airport after they did. They were off in other planes before they finally through their weary selves upon the pilot of number 17. Of course that day was an exception. A crowd was to be expected. If, however, airlines which could handle night-seeing tours want crowds they should provide methods for smoothly handling them, and they should contribute their destinations and field offers.

GROUP two were off for a well-earned jaunt both, the first time in a three-engine 6-place cabin job on a regular commercial airline. The commercial pilot was

unknown to them and they had flown the route hundreds of times themselves. Perhaps they were a little excited. The pilot kept off the air line, which is well defined with good fields, in order to fly a slightly uncommon route along the ridge of a small mountain range where practically nothing but a sparsely wooded hill stands at the point of confusion had laughably told them that the pilot had crashed on a couple of planes the week before, which made them feel particularly kindly toward him. The acts of the cabin job seemed to be merely related to the floor of the cabin. One pilot looked the empty seat in front of him and under it loose from its coverings. He didn't kick very hard either. There were no safety belts tied to the cabin structure, so that the passengers were said to slide into a confined seat in the forward end of the cabin in case of a crash. Two passengers were in shock, but the window view so tightly framed that they could not be opened. The air got pretty loud and then two other passengers got air-sick, too. The pilot made no effort to check about the luggage air. He cruised along at two thousand feet and every time the plane would hit a very bad bump he would turn around and grin at the passengers, reassuringly. I didn't seem to care very much. One woman said that she wished her passengers to what he was supposed to do along and one of the passenger pilots said that he looked pretty nervous himself and that his grin was rather forced. The other suggested that it would be given if the company would have somebody in one of the sick passengers and in the other passengers. The passengers were very speechless under such a situation. Their leader exclaimed, after fifteen minutes or so that, one rather uncomfortable and after an hour in rough air they were better. The pilot and assistant pilot didn't seem to mind this, then each had a big fat air-cushion to sit on. The pilot finally did check about at his destination. The following day the two passenger pilots were out at the airport when the same pilot took off in a smaller 3-engine ship with eight passengers in it. The other engine was cutting out as he moved it on for test at the end of the runway. He took off and just at the end of the field the same engine quit. He landed in the river bed and crashed into a small plane, which. In a minute someone was killed, but the flying crew of some of his passengers was undoubtedly permanently scarred.

On one occasion I went with two friends to an airport where a private-ride flying service conducted night flights. The flying took rather sharp, several hundred dollars' worth of new gear, new landing gear, a door, and the manager was kindly supervising the painting of a plane to his own liking. We finally caught the eye of one of the pilots. Sure, we could fly, but not with him. Why? Well, I'm not flying any blarney. I studied OXN "Sparrow" out of the field with two passengers in it. This pilot and his employees apparently differed regarding the suitability of their equipment, a matter that to say the least should be settled in private, and not before prospective customers.

The airport in question is not large and the flying service is quite a few OXN "Sparrows". The airport is closed-in to the hazy-mist district and is one of the most ideal in the country for the purpose for which it is intended—an air transport terminal. While we were talking an OXN "Sparrow" landed close to a complete stop and then took off again without taking a look. The pilot cleared the area at the end of the field by 30 to 75 ft.

Another airport was just about of him in case of a forced landing. It looked very safe to me. Two weeks later I again visited the airport. The pilot who wouldn't fly "Sparrows" was still working for the same firm, and I didn't see him by a plane of that type. The manager of the branch is not a pilot and knows little about flying a fact that speaks for itself regarding this occurrence.

ANOTHER CRASH recently occurred on one of the air transport lines. Perhaps nothing in aviation has caused more concern among the ordinary pilots, and flying and not flying people I know. They were sorry about the crash but they realize that air transport has no dangers, as does every form of transportation, and most of them have wholeheartedly accepted it despite those dangers. Their comment was dropped primarily just that the few extended operations in the period of time after the crash. They granted that search for the lost plane was necessary, but it seemed very unnecessary to them not to continue operations whether permitting, and, if necessary, risk their own personal care the right to capacity to show that they had never considered it.

Stopping operations seemed to indicate that something was wrong with the equipment or methods that needed time for readjustment. To allow the move seemed to be semi-sensational, designed to catch the sympathy of the general aviation community, and naturally established loss of revenue for the line. These opinions are not personal, I merely record impressions. Are one who deals with aircraft naturally takes some risk with human life. Air transport is a business and should be run as such.

IT WOULD SEEM that much of the above deals with private details. If one, each detail, be made directly upon the passenger who travels by air and it is therefore worthy of thought. If minor, correction should be easy.

It is suggested that placards in the cabins of passenger planes would be of interest to passengers. One such placard should give the history of the plane as an interesting way of providing of comfort that he has a history. Another might well give facts about the plane. A third might briefly and simply describe the method of taking off, cruising and landing. A fourth could describe the route, pointing out prominent landmarks. Another could give statistics on the past operations of the line—miles flown, hours, etc. Another should state that the plane was inspected prior to flight, and give some details of the inspection. Some method should be devised whereby the passenger at a glance will recognize that he is entering a plane of superior type, maintained in the best possible condition. Such cards would impress confidence by giving the given passenger knowledge that he is interested in, is indeed eager to get. They would divert his attention and lessen his fear. We worry less about things that we know a little about. It is addition, the aviation pilot or steward (or some line called crew) might have several cards covering a few simple occurrences out of the ordinary, such as passing through clouds, rain, trouble with one engine and similar matters, to be displayed should such conditions arise.

Flying being what it is, the question of pilots will ever be vital. Air transport is new, and many of the details haven't yet been worked out. There are not enough pilots to handle a number of planes now flying on air transport routes that will not hold their jobs when the present

death is overcome. Of course many of the finest pilots in the country are engaged. Good pilots develop and acquire experience and skill in a very short time, but it has been possible. It is difficult to create a pilot of young men able to cope of an accident. No other person on earth can so just how he might have acted under the same emergency conditions, and no one can exactly predict just what those conditions were. It is certain, however, that air transport operations must be directed by capable executives who are also widely experienced pilots whose decisions are beyond question. The personal element in flying is so predominant, that it appears that some system in which each pilot has a representative interest in the successful operation of the line would be worthy of trial. At present pilots are in competition with cheap equipment and poorly paid pilots in direct competition with companies that are trying to do the job in the best possible way. Higher standards must prevail, and means for increasing them provided by the air transport operator—or, eventually, it is impossible to continue to give.

WEATHER STORMS is a problem. Imperial Airways' plane on the Paris-London route was always in search with Croghan in radio phone. They receive up-to-the-minute weather reports. They can be located by radio bearings on short notice. Near our most ship radio bearings, some which might be utilized for aircraft. Along the inland routes, radio stations would be required to report weather, and the pilot would know the distance of the air line in order to obtain a satisfactory intersecting angle for the two lines of position. The initial expense and the cost of operation of such a system would be considerable, but not prohibitive. The time to provide accurate and timely reports of operation that can be readily accident-proof as possible cannot be further delayed. Certain things must be done to every air transport plane in flight is a vital step.

A number of air transport operators seem to feel that profits cannot be made until planes carrying fifty passengers or more are in use and passenger traffic produced sufficiently greater in volume than that now available. The actual and existing cost per passenger mile can be greatly reduced with larger planes and the revenue per passenger seat can be increased and still show more profit than in smaller types, provided expenses or new airports, loads can be regularly loaded. The air line that right now cannot reduce its overhead and increase the regularity of its loads by a point or two, profits are in a probable soon day to go. The public is about saturated with air accidents, and the flesh or air enthusiasm will no longer carry the financial burden of constant loss.

The average man has been sold on flying as a sport, but he is not yet when it comes to traveling that it is really here to stay. National advertising, setting forth the advantages of air travel financed by the airline operators as a group, is badly needed. It would be interesting to know how many of the passengers on any air transport line fly for business reasons and how many fly as a sport or for pleasure. It would be even more interesting to know how many passengers make one trip and then return to normal means of transportation. Every time this line occurs, air transport has failed to make a sale. Air travel must be developed into a national habit. One means of furthering this appears to be making each trip an unqualified pleasure for the passenger. The pilot should be paid not as a mere employee, but as a traveling, honored guest.

# SPECIALIZING THE PRODUCTION OF *Wooden* Parts

By JOHN F. HARDECKER

**E**XTREMELY beyond the range of the intrinsic merits of the two competing materials, wood has had to overcome the added handicap of its own condition of inferiority toward the airplane industry—a condition of which the airplane manufacturer was not unmindful, and which of necessity influenced his choice to a considerable degree. In contrast, the metal industry has been heavily successful, the airplane being largely the testing laboratory for many of the new metal alloys developed in recent years. In fact, in past years, when the volume of material used by the aircraft industry itself was an almost negligible amount, metal manufacturers have worked successfully in developing and perfecting new alloys for airplane use, realizing that if the new alloy successfully met the rigid requirements of an aeronautical usage, the desired volume of business could be readily obtained in other industries, once merit had been so established.

As the airplane manufacturer continues to make his demands, wood, the poorer material, has been steadily losing ground to metal, particularly the new light alloys. Some of this loss has been logical and inevitable, for as the aeronautic industry expanded to its present astounding proportions it was reasonable and proper that specialized

usage should exceed the range of raw materials found desirable for particular functions and manufacturing processes. However, that part of the loss which may be charged to the aircraft designer's desire to utilize a material which was of high strength weight ratio and positive in its physical properties within narrow limits, is due primarily to the lumber industry's own failure to recognize the particular needs of the aircraft industry.

While the steel industry made a close co-operative study of aeronautical requirements, and modified its own trade practices in so far as its contacts with aeronautical activity were concerned, the lumber industry as a whole was prone to stand pat on its current commercial practices and guidelines, and put the burden of proper aeronautical application of its material on the airplane manufacturer himself. The inevitable consequence was that the airplane manufacturer was somewhat lost both in wood as a raw material, and looked to the far more co-operative metal industry and more so to the source of his raw material. This shirking of the burden of proper wood segmentation from the supplier to the consumer was the primary error of the lumber industry, for it virtually demanded that the airplane manufacturer become more expert in wood

classification than the lumbermen himself, if he were to avoid the disastrous consequences of careless and inefficient grading.

**T**HAT this indifference toward the particular needs of the airplane trade has been at the bottom of the trend away from wood, is forcibly illustrated by the remarkable success of the Pusey Manufacturing Company of Hagaman, Washington, which furnishes specialized manufacturers of airplane spruce products for the aircraft industry. The specialized metal manufacturer has been a very significant factor in aeronautical development almost from the beginning, and through the gradual expansion of this activity, and the continued entrance of new metal specialists in aeronautical activity, the most productive airplane is an unstarred actuality in aviation today. This realization of true quantity production in airplane fabrication is perhaps the most outstanding accomplishment of the current aeronautical phase of technology.

That the success of the specialist in airplane fabrication is not confined to the metal trades, is amply evidenced by the organization, which for over 25 years has been functioning as specialized manufacturer for the plane-



Skidding spruce logs to the logging road

ing because such a small percentage really proved for airplane spruce stock. Even the best and most intelligent wood experts are unable to grade lumber in the rough close enough to make it economical to ship over long distances at the high freight rates prevailing, since there is too much waste at the delivered end.

Therefore the wood specialty manufacturer has been able to build up a rapidly increasing demand for finished airplane spruce manufactured at the source of the raw material. The many other spruce products made at the specialty plant through other channels for the disposition of the material which falls below the strict airplane grades, so that it is possible to credit the airplane stock for this fall-down, which otherwise might be generally worthless.

Under such conditions of specialized wood manufacture, operations go back to the logging of spruce trees, the normal operation where spruce logs are cut into lumber, the logs drying according



The big dome spruce has the added in this point for the mill



Stacking spruce after the kiln drying

*The airplane manufacturer encounters no greater problem in all his work than that centering about his initial choice of basic raw materials. While details of design may often be refined in materials other than the original ones, the initial choice between metal and wood is irrevocably built into the design. The many ramifications of this age old struggle between the*

*outstanding manufactured material and the leading natural material is really beyond the scope of this article, which is concerned primarily in presenting to the airplane manufacturer the manner whereby he can assure himself of the most practical and consistent availability of the natural material—wood—in a form best suited for aircraft construction purposes.*



















Birds over the South Pole and a report all it is published in the newspaper. It is refreshingly a topic of conversation with a number of people. How was he able to find his way back to the starting point when every direction from the Pole is the same? And does it mean other questions are current topics of conversation. "Aviation From the Ground Up" answers these questions in language which the layman can understand again. "How is it that a man of great ingenuity lives on four feet and makes his way a hard working but not a common public foot? It is a curious question of the layman. This is all carefully explained in the pages of "Aviation From the Ground Up" and "Aviation From the Sky." In addition there are chapters on Airports and Airways and some indication of business prospects for the field. All of the sections of the book are well illustrated.

In spite of its broad scope the book is compact and all of the material presented in an attractive form.

—IRVING H. SPENCER

### Flying Instruction Lore

How to Fly by Barnet Shale, The Rembrandt Company, New York City 1929 128 pages \$1.00

THE LURE OF FLAME in the aviation stages is derived from years of experience as a teacher of Navy cadets in command in the series of books by Barnet Shale, of the Rembrandt Company, New York City. The reader is taken through the different stages of a broad, new flying student in the finished pilot in paragraphs which are clear in the narrative in form that the reader can follow the progress of his evolutionary process in each step.

The Navy cadets, although the book is written for the general public for the civilian. The Navy Pilot is strongest in the safety sections used to illustrate points the author wishes to emphasize.

While not strictly a textbook on the subject of learning to fly, the book could be read with profit by any student without aviation prior or Navy cadets. It could be read with profit also by the pilot with hours in his credit for included in Lieutenant Shale's lore. It may excite curiosity of those who do not fly and when and how. In fact these questions, in spite of the fact that the right kind of emphasis should be the book worthwhile for most everyone.

It starts off in an historical note, then goes into a non-technical discussion of why an airplane flies, and how it and the engine are built. The book is devoted to chapters dealing individually with the matters of preparation for instruction, training in preparation, handling, operating, safety.

### AVIATION

January 4, 1930

standing previous landmarks with flying into aviation with it. All aspects of each entry to be well covered. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

The average aviation of aviation students can consider Lieutenant Shale's, which is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

## Abstracts and Reviews

### Racing Aircraft Tests

TESTS OF FOUR RACING TYPE AIRCRAFT IN THE TWENTY FIVE PROPELLER RESEARCH TESTS, By Fred H. Jones, NACA Technical Note No. 317

THE tests of the high-speed aerodynamic characteristics of four racing type aircraft (N-9, N-35, C-42 and C-40) were made in the 20-foot Propeller Research Tunnel.

The N-9 section was obtained by means of all of the estimates of the designers. The N-35 was obtained by similar estimates of the estimates of the designers. The N-35 was obtained by similar estimates of the estimates of the designers.

The lift and drag forces were measured at an air velocity of approximately 100 miles per hour. The lift and drag forces were measured at an air velocity of approximately 100 miles per hour.

Of the four sections, the N-9 and N-35 were the most efficient. The N-35 was the most efficient. The N-35 was the most efficient. The N-35 was the most efficient.

THE results of data presented in this report are of great interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

a bit conservative. In the last analysis, however, that progression is a matter of one opinion and one opinion. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

Particularly have these authorities dealt with the "tail" section. These chapters contain have been written for the last few years and will still be of interest to all who are interested in aviation.

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### AVIATION

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last more than corresponding expense for a motorized automobile. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

There are also not infrequently included in the book a little more than the average private owner would be able to do for himself. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

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must be remembered, of course, that these things are not to be taken as recommendations in England, for a light to a complete plan of the general type of the book.

Enough for the moment, these things compare very favorably with the English cost of operating an automobile. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

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the owner's body. A loop, which is on a level with the canopy and the parachute is opened, is suspended from the apex of the canopy by cords which extend to the attachment member. Other cords connect the canopy to the loop to maintain the canopy in a given position with respect to the loop. When cut in use the canopy is folded on the loop. It is stated that the loop forms a head which directs the air into the canopy and that the cords leading from the attachment member to the loop and then to the apex pull down upon the apex when the parachute is intended to assist in operating the canopy.

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## New Patents

### Airplane with Folding Wings

1,319,000 AIRPLANE, Charles H. McGinnis, New Providence, Conn. Filed Jan. 11, 1929. Serial No. 354,045. 5 Claims (Cl. 224-14.2)

AN AIRPLANE WING is formed in three sections—1. a rearward section connecting the fuselage and the main section; 2. a middle section; 3. a forward section. A link at each end is pivoted on the fuselage at one end and on the other end is pivoted on the end section of the wing, so that when the middle wing section is raised the links are in tension, using the rear wing section against the sails of the fuselage.

This patent is granted with a license which is considered to be a license with respect to the middle wing section, by links and crank-rocks designed to move the wing from the rear to the middle section and then to the forward section, and then to the rear section to swing against the fuselage.

It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.



### Ownership Expense in England

OWNERSHIP AND MAINTENANCE FROM THE OWNER'S POINT OF VIEW, by John Taylor, Editor of the Royal Aeronautical Society, London. Published by the Society.

IT IS NOTED BY THE author that the cost of ownership and maintenance of an aircraft is a considerable expense. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

### Propeller-Shaft Support

1,340,000 PROPELLER-SHAFT SUPPORT, Otto Maltz, Denver, Colorado. Filed Jan. 11, 1929. Serial No. 354,045. 5 Claims (Cl. 224-14.2)

IT IS NOTED BY THE author that the cost of ownership and maintenance of an aircraft is a considerable expense. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

### Parachute

1,340,000 PARACHUTE, Charles H. McGinnis, New Providence, Conn. Filed Jan. 11, 1929. Serial No. 354,045. 5 Claims (Cl. 224-14.2)

IT IS NOTED BY THE author that the cost of ownership and maintenance of an aircraft is a considerable expense. The book is a good one. It is a book of interest to all who are interested in aviation. It is a book of interest to all who are interested in aviation.

## THE BUYER'S LOG BOOK



## Pioneer Waltham Clock

A new accurate clock for timing a motorist's flight has been manufactured by the Waltham Watch Company to specifications by the Pioneer Instrument Company, 254 Lexington Avenue, Broadway, N. Y. A wood frame is the



Clock designed to show clock duration in flight

extra pair of red hands set by hand to mark the starting time. They revolve on this point and at the end of the flight the difference between the red hands and the clock hands gives the flight duration.

The clock is finished to match all other Pioneer instruments so that it is interchangeable with other instruments on the board and the dials and mountings are Army, Navy and S. A. E. standard. It is provided with a luminous dial and has a large standard hour and double time zone with the winding and setting knob placed in front and at the bottom is marked in seconds. The clock is mounted in a standard type 544 housing 7 1/2 inch and type 544 x 15 jewel movement.

## U. S. Valve Reface

A PRODUCT of many months of re-engineering and testing is the new valve refacer model VRS in new equipment by the United States Electrical Tool Company, Cincinnati. It is claimed that this refacer can be operated without difficulty by the shop help in any shop. It automatically refaces in a term of 1/2 in diameter or less, of any length with head up to 3/4 in. in diameter, and is ready to use on any angle from 15 to 90 deg. Larger circles can be obtained on special order.

The grinding wheel runs on SKP ball bearings and is driven by a pulley belt from a universal motor connected to the valve head with the special housing. The valve holding fixture is run by a separate belt from another universal motor. Both motors operate on

alternating or direct current, 60 cycles or less.

Standard equipment includes a diamond wheel dresser, outer grinding wheel support, and complete electrical equipment. It can be equipped with grinding attachment for both straight and spiral threads; modern area grinding attachment; upset grinding attachment; and attachment with automatic adjustment for grinding ends of Mallet & Pratt valve stems. A steel table with rubber rock can also be furnished.

## Weidenhoff Equipment

THE Joseph Weidenhoff Inc., Chicago, Ill., launches shop equipment in battery and electric service, including a 110 volt A.C. No. 942 supercharger for aircraft and auto engines and a new bench model 31. The supercharger designed to eliminate the loss of storage batteries is furnished with five sets of auxiliary pole shoes making it complete for charging all kinds of engines and airplanes. The equipment



Model 31 test bench

includes everything required for automobile and aircraft engines, and has additional sets of pole shoes now in stock. The weight is 150 lb and the overall dimensions 12 in high x 9 in long x 14 in wide.

Model 31 has bench for use in direct service operation—low as all usual flexible drive and an automatic centering vice which permits turning

## TRADE CATALOGS

WIRE PATENTERS. The wire products manufactured by the C. Q. Judd Corp., Southport, Conn., are described in a 32 page catalog recently published and beginning with a discussion of the various services, makes and materials for equally used. Tables of wire sizes are given, followed by illustrations and price lists for wire and screen cloth in various materials, and for other products such as dipping buckets, condenser screens, non-like filter cloth and frayed screens.

THE BOOK OF THE MONTH. A 36 page publication of the Book-Around Corporation describes in some detail how the book place is made and its use in different parts of the world under varying operating conditions. It takes up the question of timing for the various uses, giving figures on expenditure, specifications and information on accessories also included.

BATTERY CHARGERS AND SERVICES. A short bulletin was recently published covering Burton & Rogers battery chargers and services with equipment. The description this full range chargers for average storage batteries, two half watt for small service stations and two models in heavy service.

ARMATURE LIGHTING. A booklet entitled "Looking Ahead in Armature Lighting" was recently published by the Edison and General Electric Company. The booklet summarizes the part played by light in the service of airports and calls attention to important lighting problems that are likely to be encountered in the future. Various lighting installations for the airport, highway and plane station are discussed.

GENERATORS WITHOUT REMOVING THE COIL CURRENT, or distributor and offers new speed to distributor and brushes. The motor develops 24 in. and provides a speed 3-3,600 other direction through a hand lever directly connected to the hand driving vice. The distributor assembly is heated vertically, the mechanism for closing the distributor being directly connected to the motor shaft through a pulley wheel and gear running on all. Universal three jaw self check on the coil of the driving spindle clamps to distributor shaft or over gear, providing for all sizes of distributors now in use as well as new types. The rotary speed gear is mounted in horizontal position. Instantaneously under and connected with the distributor shaft.

A tachometer calibrated 100 to 4,000 r.p.m. for right or left hand rotation is connected to the motor shaft.



Parks instructor examining motor engine in student group. All Park students and engine mechanics students are given practical instruction in the big new Ford plane.

# NOW

## >>> PARKS HAS OPENED NEW OPPORTUNITIES FOR YOU!

YOU can make more money in aviation than in practically any other line of work—if you have just the right training!

Stuffed engine and engine mechanics are drawing down pay everywhere. You can do this tomorrow if you have just the right training!

Parks Air College, America's greatest aviation school, today offers you an unequalled opportunity to fit yourself for the job you want. Parks has just added an all-metal Ford six airplane, powered by three 425 h. p. Wasp engines, to its equipment. Also a new Ryan Brougham, with a 200 h. p. Wingo Whirlwind.

These planes, as well as all the other Parks equipment, are for instruction only. That means that you, as a student in our Airplane and Engine Mechanics School, will receive a sound working knowledge of the world's leading planes and engines. It means that when you graduate, you will be able to step into a profitable, skilled mechanic's job immediately.

No other school has a reputation for thorough-

(Parks Air College is America's largest and greatest aviation school.)

## PARKS AIR COLLEGE

(Division of Detroit Aircraft Corporation)

163 Missouri Theatre Building

ST. LOUIS MISSOURI

Coupon  
PARKS AIR COLLEGE  
163 Missouri Theatre Building, St. Louis, Missouri

Without cost or obligation to me,  
please send me literature only.

Name \_\_\_\_\_  
Street address \_\_\_\_\_  
City \_\_\_\_\_ State \_\_\_\_\_



Duxonian



Bantam



Ercell-Cass



Brookston



Republic R-17



Pioneer



Bomber



Victor



March 100



Steamboat Vigilance



Whitcomb



Vertical Cabin Type



Weekend IV



Duxonian



March 100



Curtiss



Weekend



Curtiss



Aero



Curtiss



Curtiss



Curtiss



Curtiss



Curtiss



Curtiss

## are CIRRUS Powered!

The American Cirrus Engine is the first of a growing family of American Engines. Let us work with your organization on power plant problems.

16 H.P.  
175 H.P.

Marysville, Mich.



Curtiss



Curtiss



Curtiss



Curtiss



Curtiss



Curtiss



Curtiss



# SPORTSMEN TO THE FRONT!

...here are two Airplane Hunters  
who rote your eye and ear



Mr. E. G. Gerlach, experienced sportsman hunter and pilot says, "The Stearman is looking well to the front in this class and is really getting going."



Mr. Wakefield and Mr. Ellis "trapped" a stearman deep in the forest after hunting.

..... When your trigger finger "itches" ... get into action " pronto" with a Stearman ship. ... R. F. Garland, oil operator and sportsman, reaches his hunting club near St. Charles Bay, Texas, from his home in Tulsa, in five and one-half hours. By train or automobile the trip would take two nights and a day. ... Newton Wakefield, one of the best known hunters of the Northwest, used to spend six days getting to big game country. In a Stearman he reaches his hunting ground in a few hours. ...

Both agree, Stearman is the ship to use ... because it is better fitted to needs of sportsman ... gets in and out of tight places ... carries heavy loads in all altitudes ... makes short landings and quick take-offs—"Stearman is the only ship we felt safe in letting hunters use over such rough country", wrote Nick Hamer of Transcontinental "Sun God" fame ... either the Business Speedster or the Junior Speedster series is adaptable to sport or commercial flight. Write or wire us.

# STEARMAN

STEARMAN AIRCRAFT COMPANY, WICHITA, KANSAS  
Division of United Aircraft and Transport Corp.



## SOCIÉTÉ LORRAINE

200 ROUTE DE BEZONS  
A ARGENTEUIL. (S.&O.)

TÉLÉPHONE  
WAGRAM 99-67

REG. COMMERCE  
SEINE N° 111-677



MOTEURS D'AVIATION

# LORRAINE



10 HP LORRAINE



15 HP LORRAINE



20 HP LORRAINE



25 HP LORRAINE



30 HP LORRAINE



35 HP LORRAINE



40 HP LORRAINE

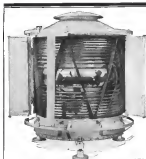


45 HP LORRAINE



50 HP LORRAINE

Marine Motors of 250 and 500 H.P.  
Automobiles 15 H.P. Sport and Touring Models



M.A.D.—Air Mail Floodlight using High Incandescence or high voltage fluorescent tubes.

## The Choice of— 78 AMERICAN AIRPORTS U. S. ARMY & NAVY— U. S.—CANADIAN, and FOREIGN GOVERNMENTS

Used, approved, and  
RECOMMENDED by

**Leading Air Transport  
Carriers and their  
Flying Personnel..**

**While a survey will show the  
World's Greatest Airports Floodlighted by B.B.T.**

**This outstanding preference for  
"AVIATION'S bad weather FLOODLIGHTS"**

should mean much to you who  
are about to buy Airport Lighting.  
Be Fair to the Pilots using Your  
Airport After Dark.

**Investigate this combination  
of features**

True 180° Beam Spread  
Complete Horizontal and vertical  
Beam Control  
Even Distribution of Illumination.  
(No high or low spots)  
Greatest Illuminated Area  
Coverage  
Maximum Intensity  
Minimum Glare

**which has made B. B. T.  
Floodlights the accepted  
Standard of Comparison.**

Our latest illustrated Catalog "Auto Airport  
Lighting" should be in your hands.  
Send for it now.

**B.B.T.**

**CORPORATION OF AMERICA**

AVIATION BUILDING  
DIVISION



B.B.T.—or blower lamp. Available with or without beam shifter.



B.B.T.—blower lamp. Available with or without beam shifter.

**YOUR AIRPORT DESERVES THE BEST. FLOODLIGHT WITH B. B. T.**

## FLYING OVER WATER

*The keenest of aviation's pleasures for men  
—who love the sea—*

**F**OR the keen yachtsman the transition from water to air is a natural one. Men who long have believed that boats afford the finest of sports have found a whole new order of enjoyment opened up for them on their first flights. Particularly flights over water.

Natural, too, is the yachtsman's instant liking for the "S-38," the Sikorsky Amphibion that flies with eight or ten passengers at 125 miles an hour. With a perfect landing "field" below you, reaching as far as the eye can see, you may fly low enough to get the full thrill of your speed. And yet the shore is always available. A touch of the convenient control drops the landing gear and you



*Yachtsmen know it is the ultimate thrill to fly over water. The "S-38" is naturally equipped, yet she is equally at home in landing or taking off water.*

are ready to taxi up the beach.

The "S-38" has a ceiling of 18,000 feet and will fly and maneuver on either engine. It has a cruising range of over 600 miles at a speed of 110 miles per hour. In brilliance of performance, in luxury and in all

around airworthiness the Sikorsky is a yachtsman's craft.

Descriptive literature will be sent on request. Sikorsky Aviation Corporation, Bridgeport, Connecticut. Division of United Aircraft & Transport Corporation.



**SIKORSKY AMPHIBION**

*This is one of a series of advertisements directed originally to advertising men in an effort to make industrial advertising more profitable to buyer and seller. It is printed in these pages as an indication to readers that McGraw-Hill publishing companies mean advertising effectiveness as well as shareholder equity.*

## Is your copy keeping step with your salesmen?

**A**N eastern manufacturer selling a product for general industrial use has advertised consistently in six McGraw-Hill publications. His sales year after year have been so satisfactory that he has readily renewed his advertising contracts.

The product is staple—one of those prosaic things that make copy writers age prematurely. A new competitive situation came up last fall that made

the copy obsolete. The advertising writer left his copy desk and turned salesman for a while. He returned with a sharpened pencil and a new viewpoint.

The new copy has been running now for several months. No change in advertising schedule! No change in sales policy! No change in product design or service! Nothing has been changed except the copy, which has become more sales-like and more humanly interesting.

**N**OW comes the president's report on sales for the first quarter. Does it not show that it pays to scrutinize copy as well as the medium that give it voice?

### THE REPORT\*

I am inclined to believe that the new type of advertising is getting the results we had hoped for. In fact it is coming much better than we had reason to expect. We are very busy in the plant at the present time and our sales for the first quarter are running 35% higher than last year, which is quite a jump. The particular class of work we were after in this advertising has necessitated our practically doubling the machinery in this department and it is now operating on a 24-hour schedule.

\*Extract from a personal letter covering several months.

# McGraw-Hill PUBLICATIONS

New York Chicago Cleveland Detroit Philadelphia St. Louis  
Greenville San Francisco Boston London



Since 1917 the  
Bellanca plane  
has been  
reliable and  
safe.

## Payload and Speed mean business

In the ordinary airplane, speed and performance are usually sacrificed to gain a greater payload, at least. In the Bellanca, remarkable speed and performance have been achieved with the highest rate of payload known to any single engine cabin plane of its class.

When the American endurance record was broken, a Bellanca biplane carried 6,300 lbs. with a two h.p. engine! Judged for combined speed, payload and quick take-off, Bellanca monoplane have five times won the Aviation News and Country Club Trophy and three times the Detroit News Trophy—America's most important efficiency contests. Again in the recent Ford Reliability Tour, Bellanca plane swept their class to an unprecedented victory.

The new 1959 Bellanca Pacerplane achieves a speed of 141 m. p. h., and a payload (including pilot) of

1,250 lbs.—ten passengers and 200 lbs. of baggage. The Bellanca Freighter easily carries a payload of 2,500 lbs.—the only type licensed by the U. S. Department of Commerce for a useful load greater than its own empty weight. Here is a combination of performance and payload that means business! The new Bellanca have inherited all the perfected features and the well-known ability of their famous predecessors—a phase of luxurious appointments, rare comfort, unexcelled safety, and extraordinary flying qualities.

If it's a business getting serious that you want, get the facts on the Pacerplane!

**Bellanca Aircraft Corporation**  
New Canaan, Delaware

Bellanca Agents at the following points will be pleased to accept your inquiry in every possible way.

Baltimore: Hines, Inc., 111 E. Calvert St.  
Baltimore: Bell & Howell Photo Lab.  
New York

Bellanca Aircraft at Dallas, Ind., 1951 St.  
Columbus: St. West, 1000 E. 10th Ave.  
Chicago: Bell & Howell Photo Lab.

Easton: Wright, 1000 E. 10th Ave.  
Philadelphia: Bell & Howell Photo Lab.  
Pittsburgh: Bell & Howell Photo Lab.

St. Louis: Bell & Howell Photo Lab.  
St. Paul: Bell & Howell Photo Lab.  
Tampa: Bell & Howell Photo Lab.

Seattle: Bell & Howell Photo Lab.  
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Seattle: Bell & Howell Photo Lab.

# BELLANCA



**"We need a plane—  
even our chief engineer travels 35 days a year"**

TRAVEL in business is a necessity.

A survey conducted by one manufacturing company showed that even the chief engineer traveled 35 days each year—time that cost a total of \$2,804.39 in salary, and was lost for productive work while he was away from his desk. In addition, every sales representative, from president to assistant sales manager, averaged more than 4 days a month on the road—time that could have been reduced to a minimum by the use of fast air transportation.

The road travel bill for this firm last year—including salaries and other expenses—was \$22,748.00. As a means to economize, one of the executives advised the president to purchase an airplane—and after a thorough investigation the company bought a Ryan. Result? While the Ryan cost \$15,750.00, it was



actually paid for itself in one year in time saved. And, more significant still, the company has since secured more business, made more contracts and enjoyed greater prestige than at any other period in its history.

In this country of vast distances and dollar-and-centa valuations on time, Ryan airplanes are business assets. With a speed of 150 m. p. h. and a cruising range of 700 miles they save countless hours and escape every successor's field of anxiety. And a Ryan costs less than two cars to operate—with allowance for writing off the entire depreciation in the first year.

Let us tell you more about Ryan airplanes—show you how they can completely solve the transportation problem for your company.

BUILDERS OF LIGHTBULBS



"SPIRIT OF ST. LOUIS"

**DETROIT AIRCRAFT**

UNION TRUST BUILDING, DETROIT

WESTERN OFFICE: ROOSEVELT BUILDING, LOS ANGELES

SPAN AIRCRAFT CORPORATION  
LOOKED AIRCRAFT CORPORATION  
PARKS AIRCRAFT CORPORATION  
AIRCRAFT PARTS COMPANY, INCORPORATED

PARIS AIRCRAFT CORPORATION  
BATHMAN AIRCRAFT CORPORATION  
BATHMAN AIRCRAFT CORPORATION  
AIRCRAFT DEVELOPMENT CORPORATION

MAIRIE AIRCRAFT CORPORATION  
BOSCH AIRCRAFT CORPORATION  
GLIDERS, INCORPORATED  
DETROIT AIRCRAFT EXPORT CORPORATION

# Thoroughbreds . .



ARMY P-6 HAWK—top speed over 180 miles per hour. The Air Corps' high-performance fighter powered by the reliable Conqueror engine.

CURTIS CONQUEROR ENGINE—12 cylinders, geared, liquid-cooled, Vee type engine that sets new performance records for fighting planes.



## Curtis' Engineering Leadership Has Developed Two Thoroughbreds P-6 Hawk and the Famous Conqueror Engine

THE Curtiss P-6 Hawk is a thoroughbred. It's the latest development of the distinguished Hawk series of fighting planes.

The outstanding performance of the Hawks in the service of the crack fighting squadrons of the Army and Navy has become famous throughout the world.

Now refined in design and powered with the liquid-cooled, geared Curtiss Conqueror Engine of 625 horsepower, the new P-6 Hawk sets new standards. Its top speed is in excess of 180 miles per hour. The Hawk rightfully takes its place as the leader in performance and maneuverability

in the Air Corps. The Conqueror Engine, with which the Hawk is powered, is also a thoroughbred. Its predecessor, the famous Curtiss D-12 Engine, has been the high-performance aircraft engine in this country and abroad since its introduction.

The Conqueror develops 625 horsepower delivered through a geared propeller. It is exceptionally compact, low in frontal area and inherently reliable. It has been proven a truly remarkable engine by its long use in the government air service branches.

Curtiss Aeroplane & Motor Co., Inc. Office, Garden City, Long Island, N. Y. Factories, Garden City and Buffalo, N. Y.

**CURTIS AEROPLANE & MOTOR CO., INC.**

# THE KITTYHAWK

Some experienced pilots have taken the Kittyhawk into the air . . . and failed. Others have succeeded . . . but even they have been unable to hold the Kittyhawk in a spin for more than half a dozen turns.

With a landing speed of 28 miles per hour . . . with a wheel tread of seven feet two inches . . . the Kittyhawk is particularly adept at getting in and out of small rough fields . . . In fact the Kittyhawk has all the safe flying characteristics which makes it a very desirable plane for the student and inexperienced pilot . . . and yet the expert will find that it will out-perform most every plane in its class.

Write to us . . . we shall be glad to send you further information about the Kittyhawk.

## THE VIKING FLYING BOAT COMPANY

89 Shelton Avenue, New Haven, Connecticut

Miami Hanger—Municipal Airport, Casaway Island

COMPANION  
TO THE  
VIKING  
FLYING BOAT



### SPECIFICATIONS:

Engine—Beech E-5  
—Beechcraft (Last Certificate No. 348)  
Length overall 28 ft. 11 in.  
Wing span 35 ft. 5 in.  
Wing high 10 ft. 10 in.  
Wing low 10 ft. 10 in.  
Wing area 153.4 sq. ft.

#### Kittyhawk Model B-4

Weight empty 1050 lbs.  
Landed load 1500 lbs.  
High speed 100 m. p. h.  
Cruise speed 80 m. p. h.  
Landing speed 28 m. p. h.  
Climb 1000 ft. p. m.

Wing area 153.4 sq. ft.  
Wing high 10 ft. 10 in.  
Wing low 10 ft. 10 in.  
Wing area 153.4 sq. ft.

## Move the Bench Not the Job



Push and Pull—Benching

## "Hallowell" Semi-Portable Work-Bench of Steel

A Semi-Portable Work Bench on two casters—one you can trouble around from place to place, from job to job—how does this strike you?

Before the "HALLOWELL" Semi-Portable or Steel work bench was a fixture—once its position there is required.

Start us with the "HALLOWELL"—grab the first day and trouble Bench and all to where you want it.

Let go and the "HALLOWELL" lands on its own place rest and become, at once, rigid as a rock.

And when the job is finished, roll the "HALLOWELL" away and leave it where least in the road, so floor and wall space can be used to best advantage.

Otherwise the construction is just like the new famous line of

## "Hallowell" Steel Benching

Be Bette 336, 415, and 456—Yours for the Asking.

# STANDARD PRESSED STEEL CO.

BRANCHES  
BOSTON  
CHICAGO  
DETROIT

JENKINTOWN, PENNA.

BOX 538

BRANCHES  
NEW YORK  
SAN FRANCISCO  
ST. LOUIS











[illegible]

# Business Wants

**T**HE Searchlight Section of this paper represents a meeting place for men and concerns who have immediate business "wants" to fill—the section covers

Apprentice Wanted	Five Salesmen Wanted
Agents Wanted	Gilbert House For Rent or Wanted
Berlin and Perchville	Partners Wanted
Business For Sale or Rent	Travel Agencies
Civil Service Opportunities	Partners For Sale
Condoms For Sale	Partners Wanted
Daily Rates For Room or Board	Partners Wanted
Edinburgh	Private Bar For Sale
Employment Agencies	Representatives Wanted
Employment Service	Submarine Available
Foreign Travelers	Submarine Wanted
For Sale	Swiss House With View
For Sale	Sub Particulars Wanted
For Sale	Teaching
For Sale	Traveling Work Wanted
For Sale	Work Wanted

# "SEARCHLIGHT"

**FROM FACTORY TO USER!**

The first and only of its kind! Supreme's new propeller is the first to be made in America by a company that has been in the propeller business since 1914. It is the only propeller that is made in America by a company that has been in the propeller business since 1914.

**SUPREME**

Write or wire for details and discount!

**Supreme Propeller Co.**

**WICHITA, KANSAS**

1938

**SEAMLESS  
STEEL TUBING**  
*All Sizes & Grades*  
Wholesale prices for seamless  
tubing of any quantity.  
Mill discounts for scheduled  
production requirements.  
**SERVICE STEEL COMPANY**  
800-541-5511  
1470 Lincoln St.  
LANSING, MI 48201  
248-462-5511

**PLYWOOD**  
FOR  
**AERONAUTICAL  
CONSTRUCTION**  
**New Jersey Veneer Co.**  
Paterson, N. J., U. S. A.

**BURGESS**  
**DRY CELL BATTERIES**  
 Have Passed the Test  
 under all kinds of conditions  
 for  
 Aircraft, Boats, Ignition, Flash-  
 light and low voltage lighting  
**BURGESS**  
**BATTERY COMPANY**  
 General Sales Office, Chicago

AVIATION  
January 4, 1939

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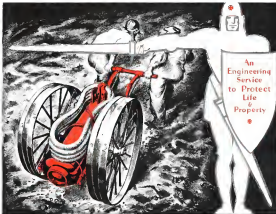
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At 7:00 they were nearing Gould’s geological party struggling along the trail beneath. “*Motors fine.*” By midnight they had covered seven hundred miles and the Pole lay ninety miles beyond. “*Motors fine.*”

At 1:00 Byrd’s calculations indicated they had reached the Pole and were flying high for observation. Plane and crew in good shape. “*Motors fine.*” They crossed the Pole . . . flew beyond it . . . around it . . . and then turned North. “*Motors fine.*”

At 4:33 they landed at the Mountain Supply Depot, re-fueled and were on their way again at 6:00. “*Motors fine.*” At 8:00 they crossed again the great crevasses, flying well. “*Motors fine.*” At 10:10 Balchen landed at Little America after flying 1600 miles in 17 hours and June signed off. “*Motors fine.*”



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